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Highlights

- A URANS CFD simulation approach to investigate VAWT fluid dynamics is presented
- The focus is on the analysis of the performance increments observed in skewed flow
- The 3D CFD model is able to capture the unsteady 3D effects such as dynamic stall
- New light has been shed on understanding the origin of the performance increments
- Gain mainly due to the downwind part of the rotor being less disturbed by the wake

3D URANS Analysis of a Vertical Axis Wind Turbine in Skewed Flows

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Abstract

The article demonstrates the potential of an unsteady RANS 3D approach to predict the effects of skewed winds on the performance of an H-type vertical-axis wind turbine (VAWT). The approach is validated through a comparison between numerical and experimental results for a full-scale Darrieus turbine, demonstrating an improved prediction ability of 3D CFD with respect to both 2D CFD and semi-empirical models based on the double multiple stream tubes method. A 3D URANS approach is then adopted to investigate the power increase observed for a straight-bladed small-scale turbine in a wind tunnel when the rotational axis is inclined from 0° to 15° from the vertical. The main advantage of this approach is a more realistic description of complex three-dimensional flow characteristics, such as dynamic stall, and the opportunity to derive local blade flow conditions on any blade portion during upwind and downwind paths. Consequently, in addition to deriving the turbine overall performance in terms of power coefficient, a better insight into the temporal and spatial evolution of the physical mechanisms is obtained. Our principal finding is that the power gain in skewed flows is obtained during the downwind phase of the revolution as the end part of the blade is less disturbed by the wake generated during the upwind phase.

Keywords: vertical axis wind turbine, offshore, skewed flow, dynamic stall, CFD

1. Introduction

The pursuit of reducing the cost of offshore wind energy in deep waters has led to a re-emerging interest in vertical-axis wind turbines (VAWTs) for floating applications due to apparent advantages over conventional horizontal-axis wind turbines (HAWTs) [1, 2, 3]. In parallel, there is a resurgence of interest in VAWTs as a promising alternative to HAWTs also for small-scale electric power in urban areas [4].

VAWTs are characterised by some significant advantages: the ability to capture wind from any direction without a yaw control mechanism, low noise, compact design, simpler

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4 access, installation, maintenance and repair (since the gearbox and drive train components
5 can be located at ground level rather than at the top of the tower as for HAWTs).

6 Furthermore, although VAWTs generally have a lower aerodynamic efficiency, there is
7 some evidence that VAWTs can be positioned closer together in a wind farm giving a higher
8 power density due to lower wake interference. Furthermore, counter-rotating VAWTs in
9 close proximity have been shown experimentally [5] and numerically [3] to have a mutually
10 beneficial effect on power production.

11
12 It is perhaps the offshore environment that has attracted the greatest interest for VAWTs
13 because of several inherent attributes that offer advantages with respect to HAWTs, partic-
14 ularly the scalability and low over-turning moments with better accessibility to drive train
15 components [6, 7].

16
17 A correct prediction of the turbine performance in skewed flows is very important for
18 both micro generation in the built environment and large-scale offshore applications [8],
19 since urban winds have noticeable vertical components (micro generation), and waves and
20 unsteady wind speeds will induce pitch/roll motions to the turbine axis and therefore a
21 periodic tilt angle with respect to the vertical (offshore floating wind turbine). HAWTs and
22 VAWTs exhibit completely different behaviours in skewed flows. Theoretical studies and
23 experimental measurements have shown that the power output in a skewed flow is reduced
24 for an HAWT [9]. This trend is mainly due to a reduction in the effective swept area
25 (i.e. the area perpendicular to the oncoming wind direction). Consequently, initial studies
26 have indicated that HAWTs can suffer severe performance losses when installed on floating
27 support structures [10]. On the other hand for VAWTs, depending on the configuration, the
28 performance degradation due to skewed flow is generally lower and for H-VAWT (VAWT with
29 two vertical blades connected to the central tower through one/more arms) configurations the
30 power coefficient may even be enhanced. Experimental investigations on H-VAWT turbines
31 have demonstrated an enhanced performance for relatively small tilt angles (up to $25^\circ - 30^\circ$)
32 depending on the shape of the rotor, whereas a reduced performance is produced by higher
33 skew values [11]. This benefit might be explained by the fact that VAWT blades sweep out
34 a cylindrical surface, as opposed to a planar surface for HAWTs. As a consequence, during
35 misaligned flow operations, the swept area of the turbine is increased.

36
37 To account for skewed flow effects on VAWT performance, the efforts of researchers have
38 focused on the implementation of semi-empirical corrections in blade element momentum
39 models [12, 13]. To the authors' knowledge, there are no studies in the literature presenting
40 a 3D CFD (Computational Fluid Dynamics) study on inclined VAWT performance. The
41 present study is aimed at contributing to a better understanding of the physical processes
42 that result in the measured performance enhancement reported for VAWTs in skewed flows.

43 44 45 46 47 48 49 50 51 **2. Numerical approach**

52
53 The two wind turbine configurations analysed in this work are:

- 54
55 • the SANDIA 17m-diameter Darrieus-Type VAWT - used to validate the CFD mod-
56 elling approach that has been adopted. It has an aspect ratio (height to diameter ratio)

Mesh	No. nodes on blade	No. elements	Wall distance
1	620	301163	1.6 E-05
2	1240	464845	1.6 E-05
3	1920	700292	1.6 E-05

Table 1: Details of the 2D grids

equal to 1.02, two blades with a NACA 0015 aerofoil section and a constant chord of 0.612 m. Rotational speed is fixed at 38.7 *RPM* (rotational speed, in revolutions per minute) for aerodynamic force measurements and 42.2 *RPM* for c_p measurements [14];

- a two bladed H-Darrieus VAWT with NACA 0018 aerofoil sections and constant 0.08 m chord. The rotor height is 0.5 m and diameter is 0.755 m. This configuration was used in wind tunnel tests to study the influence of skewed flows [4] allowing direct comparison with CFD predictions.

For all the simulations, two different grid levels have been adopted: a fixed sub-grid with the external dimensions of the flow domain, and a dynamic sub-grid that includes the VAWT geometry and allows a relative motion with respect to the fixed grid. This grid arrangement utilises the sliding mesh technique [15] and allows the simulation of the rotating motion of the wind turbine with a steady RANS (Reynolds-averaged Navier-Stokes equations) or URANS (unsteady RANS) analysis.

The CFD process was validated for the SANDIA National Labs configuration using both 2D and 3D grids. The 2D mesh adopted is a hybrid structured-unstructured mesh with around 460 000 elements. The flow domain external dimensions were $37D \times 25D$, and the wall distance from the first layer of cells is set at $1.6 \cdot 10^{-5}c$ with y^+ (dimensionless wall distance) < 1 , where D is the diameter of the turbine and c is the blade chord length. The 3D grid contains around 2 800 000 elements with flow domain external dimensions of $15D \times 5D \times 4D$ and a wall distance equal to $1.9 \cdot 10^{-4}c$, resulting in $y^+ < 5$.

For the analysis of the rotor in a skewed flow, only a 3D approach was considered and the mesh contains around 1 800 000 elements. The external dimensions are $14D \times 30D \times 12D$ and the wall distance is $1 \cdot 10^{-4}c$, necessary for an averaged $y^+ < 5$, following a mesh sensitivity analysis performed in order to verify if the grids adopted were sufficiently fine to resolve the primary flow features. Figure 2 shows the analysis for the 2D model for the 17m Darrieus VAWT and the 3D model for the small H-Darrieus VAWT; the 3D model of the SANDIA turbine has been excluded from this analysis due to the high number of cells of the domain, but the mesh adopted is similar to that used by Howell [16] and Zhang [17].

2.1. 2D and 3D mesh sensitivity analyses

For all the cases presented, both for the 2D and the 3D models, mesh sensitivity analyses have been conducted, to assess the adequate mesh element number able to resolve the flow. As regards the 2D model, three meshes were analysed (coarse, medium, fine) as shown in 1, keeping the wall distance fixed at $1.6 \cdot 10^{-5}$ with $y^+ < 1$.

Mesh	No. elements	Nodes around blades	Nodes spanwise	Wall distance
4	1724930	84	200	1.6 E-04 chords
5	2672993	186	300	1.6 E-04 chords

Table 2: Details of the 3D grids for H-Darrieus turbine

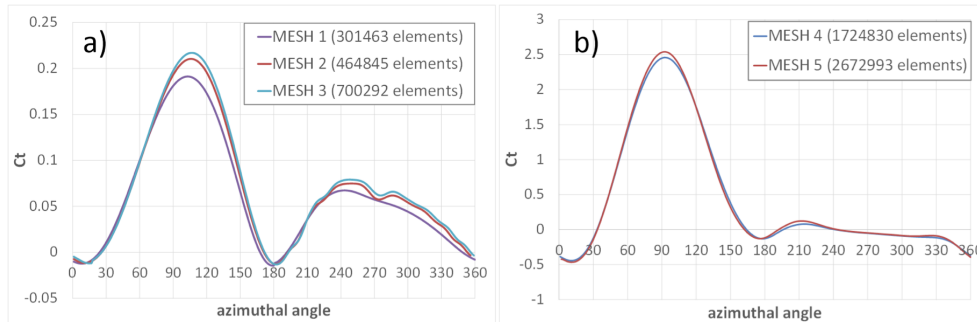


Figure 1: Mesh sensitivity analysis for 2D model of 17-m Darrieus-Type VAWT (a) and for the 3D model of H-Darrieus VAWT (b)

The 3D mesh independence study was performed only for the skewed flow case. The same philosophy was then applied to the SANDIA case. For the 3D model of the turbine in skewed flow, two grids have been considered (4 and 5), as showed in Table 2. In general a larger number cases are analysed in order to conduct a sensitivity analysis, but in this case we considered the previous work done in this area (see cited articles). All the 3D meshes were built considering a wall distance equal to $1.9 \cdot 10^{-4}c$, resulting in $y^+ \approx 5$.

The results of these analyses are showed in Figure 1, where the C_t (tangential force coefficient) values are compared.

As regard the 2D analysis, the difference between mesh 2 and mesh 3 is minor, and mesh 2 was adopted, in order to save time and computational resources without substantially affecting accuracy. The same conclusion can be derived from the second graphs, showing a small gap between the two grids. So that MESH 4 was used for the 3D analysis. Furthermore, this meshing philosophy was subsequently adopted for the SANDIA turbine case.

2.2. CFD solver approach

The CFD software used for the present analyses is FLUENT v.15, developed by Ansys Inc. This study is based on the URANS implicit model, and turbulence is modelled using the $k-\omega$ SST (Shear Stress Transport) model. This turbulence scheme was adopted because of its aptitude in cases involving high adverse pressure gradients and therefore smooth surface separations [18]. The air was considered as incompressible since the different cases studied did not exceed a local Mach number greater than 0.3.

The setup settings for the simulations are showed in Table 3. The simulations are done using a First Order Implicit scheme for all the variables of the spatial discretization in the first revolutions, after that increase the order of the schemes considered to prevent the model from numerical diffusion errors. The convergence criteria were set at $1 \cdot 10^{-5}$ for all residuals.

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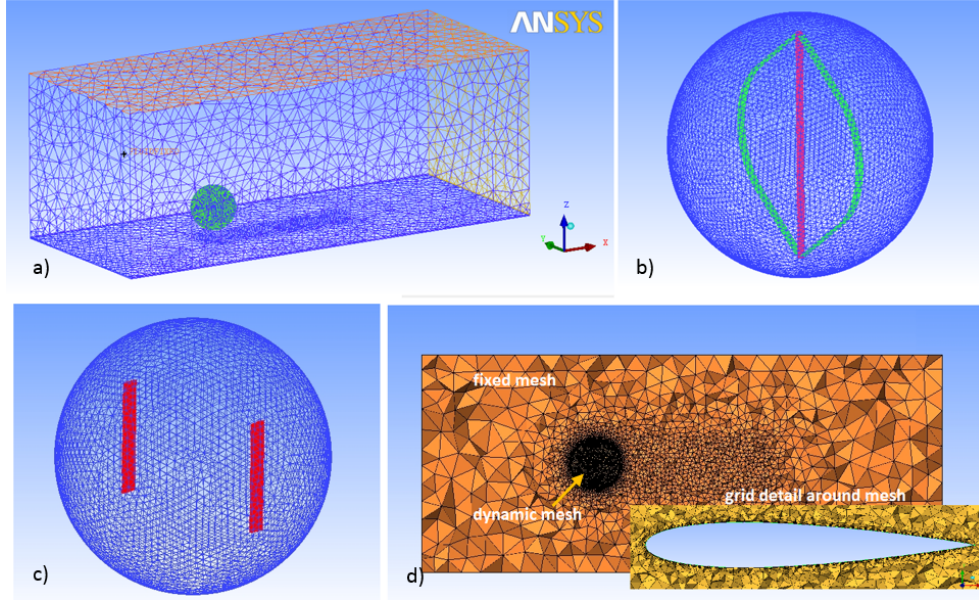


Figure 2: (a) External domain, (b) details of 3D grid for SANDIA 17m Darrieus-Type wind turbine, and (c) of the H-Darrieus Type Wind Turbine, (d) volume mesh

Solver	
Type	Pressure-based
Time	Transient
Solution methods	
Pressure-Velocity coupling	PISO
Spatial discretization	
Gradient	Green-Gauss node based
Pressure	Standard
Momentum	Second order upwind
Turbulent kinetic energy	Second order upwind
Specific dissipation rate	Second order upwind
Modified turbulent viscosity	Second order upwind
Transient formulation	
Second order implicit	

Table 3: CFD simulations: setup settings

3. Results

3.1. Verification of the CFD model

This paper focuses on two wind speed operating conditions: $TSR = 4.6$ (Tip Speed Ratio) (low-wind speed case) and $TSR = 2.33$ (high wind speed case), for the SANDIA 17m-diameter test case. Since chordwise pressures for this case were measured at the equatorial station of the turbine blade, it is expected that the local flow at this station is 2D and therefore that 2D CFD simulations give a good approximation of the flow. Normal and tangential force coefficients were obtained by integrating the chordwise pressure distributions as reported by Akins [14]. The normal and tangential force coefficients c_n and c_t are defined as:

$$c_n = \frac{F_n}{\frac{1}{2}\rho U_\infty^2 c} \quad (1)$$

$$c_t = \frac{F_t}{\frac{1}{2}\rho U_\infty^2 c} \quad (2)$$

where F_n and F_t are, respectively, the normal and tangential force on the blade per unit of span [N], ρ is the air density [kg/m^3], and U_∞ is the free stream wind velocity [m/s].

Predicted normal and tangential force coefficients from both 2D and 3D CFD simulations are compared against the experimental data in Figure 3 (equatorial plane of the rotor). The azimuthal angle of the blade θ is measured such that $\theta = 0^\circ$ at the beginning of the upwind path. These figures show good agreement between the 2D and 3D CFD models and the experimental data. The shapes of the normal and tangential force coefficient curves reproduce the measured trends well from a qualitative point of view, though there are some differences in peak values at corresponding azimuthal angles.

As would be expected, the predicted normal force coefficient data are in closer agreement with measured data than the tangential force coefficient data, with a maximum relative difference of 15% over the upwind cycle of the revolution ($70^\circ < \theta < 130^\circ$), whilst for the downwind cycle any differences (with the 2D predicted data) are small. Both 2D and 3D CFD models underestimate the tangential force coefficient at all blade positions, resulting in a maximum relative difference of $\sim 55\%$ over the downwind cycle.

Figures 3c and 3d show 2D CFD results obtained using different time steps to assess how they affect results at the higher wind-speed condition ($TSR = 2.33$), where flow separations and dynamic stall can influence performance. Because of these more complex and dynamic features, a time-step sensitivity analysis was performed using; 1° , 0.5° and 0.25° of azimuthal angle for each time step. Generally, the 2D simulation tends to over-predict tangential and normal force coefficients at this flow condition. Measured data at this condition indicate that flow separates and subsequently reattaches on the blade surface between 80° and 130° blade azimuth positions, with a potential increase in both the normal and tangential force coefficients due to dynamic stall over the final quarter of both upwind and downwind cycles. The CFD models give good agreement when the flow is attached, but separation appears to be delayed due to a higher blade azimuth angle (and therefore blade angle of attack).

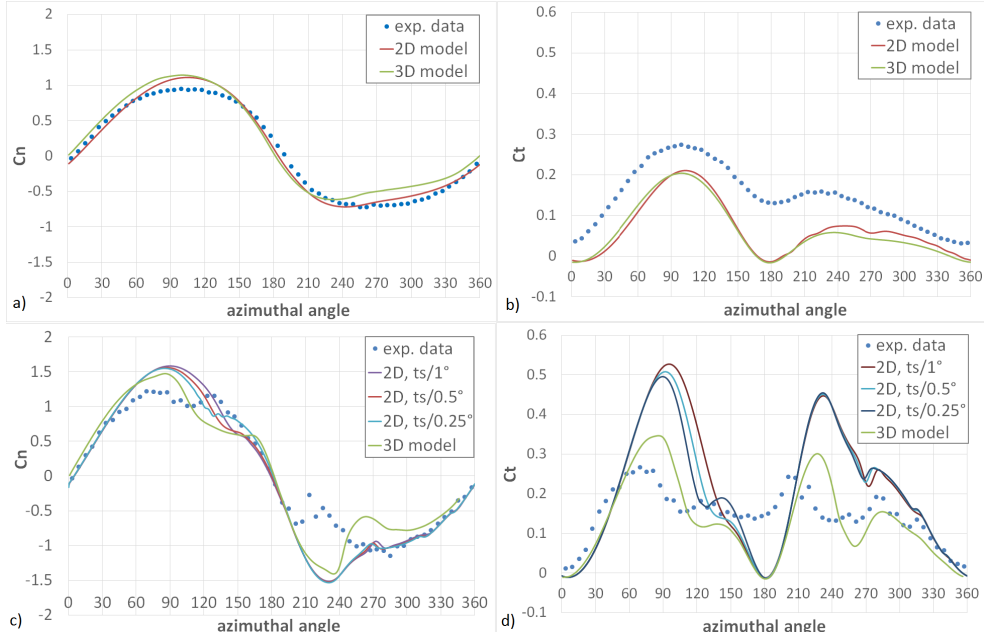


Figure 3: Normal force coefficient and tangential force coefficient, calculated at $TSR=4.6$ (a,b) and at $TSR=2.33$ (c,d)

Consequently the model overestimates the peak normal force coefficient in the first half of the revolution, with a maximum relative error of $\sim 28\%$ for the 2D model. If a smaller time-step is adopted, the separation of the flow occurs at a smaller azimuth angle but does not have a significant effect. The CFD models also predict a delay in the occurrence of stall over the downwind cycle, resulting in a big difference with experimental data. The 3D model gives a better prediction of the flow behaviour and of the aerodynamic forces on the blade, particularly in terms of tangential force coefficient. One advantage of the 3D approach could be a better estimation of dynamic stall and also lag effects due to the unsteady flow. The 2D CFD models cannot fully describe the complex dynamics of the flow whilst the 3D CFD model can take into account the near and far field flow developments surrounding the real geometry of the wind turbine, as also shown by Howell [16].

As regard the discrepancies between the experimental and numerical estimations of c_t , the following should be considered. As very well known, since integrating pressure distributions to get the tangential force is much less accurate than integrating them to get the normal force, particularly if the pressure data is sparse, the accuracy of c_t is in general lower than the one of c_n . Furthermore, it is shown that the CFD approach adopted predicts well the trends at low speed, as expected when the flow is fully attached. Also, the power is well predicted at this condition, which depends on the tangential force and so suggests that the predicted tangential force is reasonably accurate. At the higher wind speeds the flow is much more complex, and so inaccuracies in predicting the dynamic flow are expected. Anyway, the overall power is still well predicted, which indicates again that tangential force estimation is reasonably accurate.

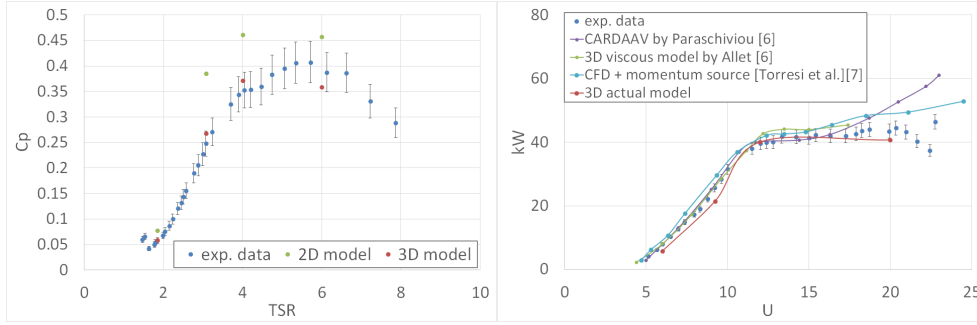


Figure 4: c_p and power curve as function of TSR

Figure 4 compares the predicted and measured coefficient of power (c_p) and power production curves for the SANDIA 17m Darrieus turbine. C_p is defined as:

$$C_p = \frac{Power}{\frac{1}{2}\rho U_\infty^3 DH} \quad (3)$$

where H is the blade height [m]. Due to the better prediction of the tangential force coefficient at higher wind speeds, the 3D CFD results are in closer agreement with experimental data and are generally within 10%. The 2D CFD model over-predicts power at high wind speeds, as would be expected from the previous analysis. Figure 4 also compares the data with those predicted using BEM (Blade Element-Momentum) approaches by Allet [19] as well as another CFD approach [20]. The advantage of the CFD approach becomes clearer at high wind-speed conditions, where greater accuracy in predicted power is achieved.

3.2. Rotor in skewed flow

Since the purpose of this work is to determine the influence of changes to the inclination of a VAWTs' rotational axis relative to the flow direction on the rotor performances, a 3D CFD model is necessary. The two bladed H-Darrieus VAWT configuration was used, with measured wind tunnel test data available for a straight and inclined rotor [4]. A preliminary analysis in order to assess the best turbulence model scheme was performed for this wind turbine, without skewed inflow. This analysis considered two of the most referenced turbulence models for cases including strong adverse pressure gradients: k- ϵ RNG (Re-Normalisation Group) and k- ω SST. The comparison was based on the calculation of the power coefficient C_p at different values of TSR .

All results refers to the seventh revolution, where the difference between the average total torque coefficient C_m and the one of the previous revolution is within 1%. SANDIA turbine has been analysed at two different wind conditions:

- $TSR=4.6$ (corresponding to a free stream velocity of 7.345 m/s) with a time scale corresponding to 1° /time-step (0.0043 s)
- $TSR=2.33$ (corresponding to a free stream velocity of 14.5 m/s) with different time scales corresponding to 1° /timestep (0.00431 s), 0.5° /time-step (0.00216 s) and 0.25° /time-step (0.00108 s)

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4 As regards the simulations of the H-Darrieus turbine, the free-stream velocity was fixed at
5 7 m/s and all the results refer to a time scale equal to 1° /timestep at each TSR considered.

6 Figure 6 shows good agreement of the $k-\omega$ SST model with wind tunnel test data, though
7 the $k-\epsilon$ RNG scheme gives poor results, particularly at low wind speeds. The difference
8 between the $k-\omega$ SST model and the measured data is within 10%, except at the highest
9 wind speed. This difference between these two models can be attributed to an overproduction
10 in wall shear stresses by the fully-turbulent $k-\epsilon$ RNG model, which led to a smaller averaged
11 torque value. In fact, As showed in Figure 5, $k-\epsilon$ RNG model gives higher values of wall
12 shear stresses than the one calculated by the $k-\omega$ SST model. These generate forces which
13 oppose the motion of revolution of the turbine, resulting a lower averaged torque coefficient
14 and thus a lower power coefficient.

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17 Both test cases have demonstrated that the $k-\omega$ SST model can simulate the power
18 production of a VAWT with reasonable accuracy. Consequently, the $k-\omega$ SST turbulence
19 model was adopted for quasi-static simulations of the rotating H-VAWT configuration at
20 fixed inclination or tilt angles. The inclination angle Φ was varied in 5° increments from 5°
21 to 15° . Higher inclination angles were not considered since it is assumed that the H-VAWT
22 is installed on an offshore floating platform and these structures are designed not to exceed
23 a limit of $\pm 15^\circ$ oscillations about the vertical axis [8].

24
25
26 Figure 7a refers to the ratio of the maximum value of c_p at different tilt angles normalized
27 by the maximum value at 0° skew angle; figure 7b refers to the ratio of TSR at which the
28 maximum c_p is reached at different values of Φ , normalized by the maximum value at $\Phi = 0$.
29 The c_p ratio has been considered rather than the c_p since only the first was available in the
30 report considered. The CFD results are also compared with measured wind tunnel results
31 and results from a BEM model developed by Mertens [4]. The BEM model under-predicts
32 the power in a skewed flow, while the CFD results are in good agreement with the wind
33 tunnel measurements.

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35
36 This means that the present model can well describe the behaviour of the flow acting on
37 the blade sections, such as the rise of dynamic stall and the effects of the wake generated
38 by the previous blade on the oncoming one. The comparison also reveals that the Mertens
39 model does not agree very well with the experimental data. The main reason is due to the
40 DMST (Double-Multiple Stream Tube) model not including the semi-empirical corrections
41 that account for the dynamic stall. Indeed, this consideration is linked to the results shown in
42 Figure 7b. It is evident that, at a higher TSR ratio, which implies higher rotational velocity
43 of the wind turbine (since the inlet velocity is fixed at 7m/s), and at higher tilt angles, the c_p
44 ratio values are closer to the one obtained in the wind tunnel tests corresponding to the same
45 tilt angle. The combined effects of the increased rotational velocity and the augmentation
46 of the tilt angle lead to a smaller influence of the dynamic stall and so a better matching
47 between the Mertens model and experimental data. However, also shown in this graph, the
48 present model matches the TSR ratio measurements better than the Mertens model.

49
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51 As seen from figure 7, the c_p increases with Φ due to the increases of the blade tangential
52 force coefficient. The normal and tangential force coefficients predicted at the equatorial
53 plane of the rotor are shown in Figures 8a and 8b respectively. As regards the normal forces,
54 small changes are seen in the first half of the revolution, where the maximum difference
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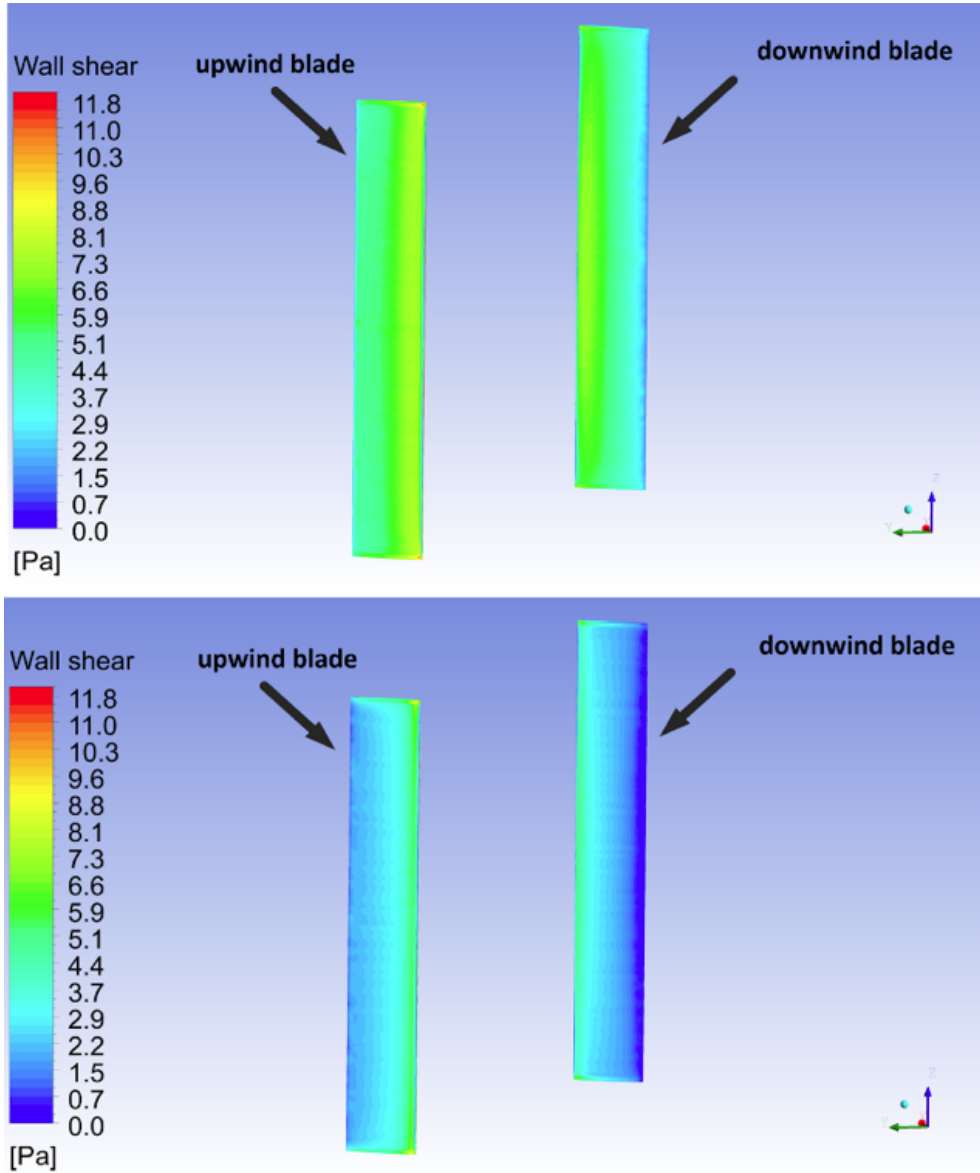


Figure 5: Wall shear on blades for $k-\epsilon$ RNG model (top) and $k-\omega$ SST model (bottom)

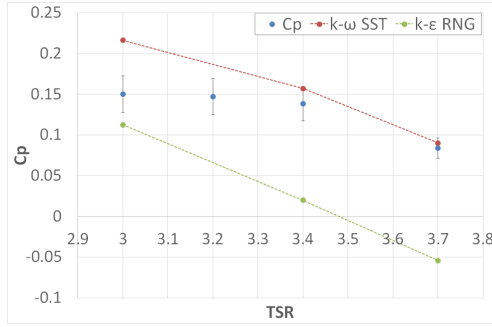


Figure 6: c_p curve for the H-Darrius experimental wind turbine. Experimental data from wind tunnel test by Mertens [4]

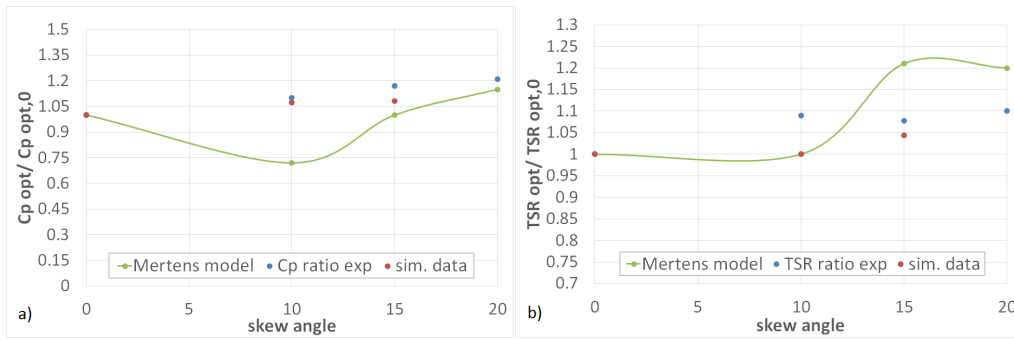


Figure 7: c_p ratio and TSR ratio for H-Darrius Turbine at different tilt angles. The ratios refer to the maximum values of the variables reached at different tilt angles normalized by the maximum value calculated at 0 tilt angle. From wind tunnel test by Mertens [4]

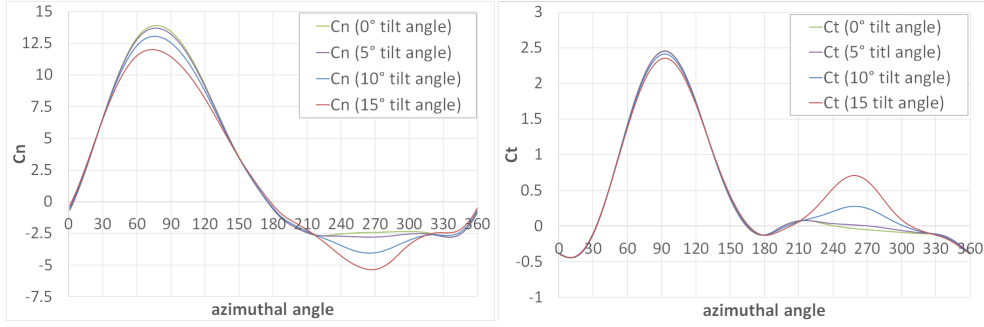


Figure 8: Normal force and tangential force coefficient as a function of the azimuthal angle, at different tilt angles

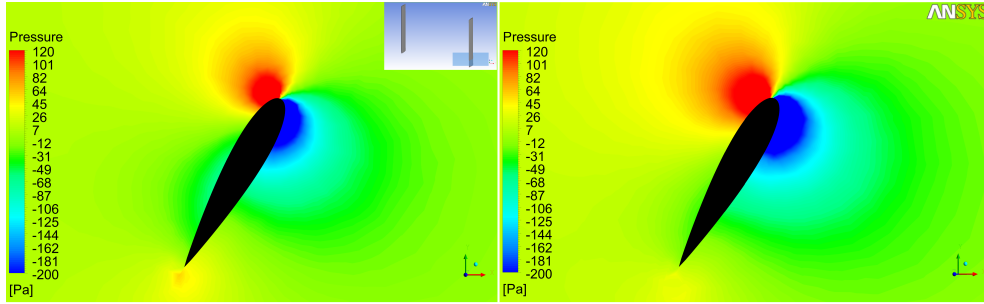


Figure 9: Pressure field calculated at $\Phi = 0^\circ$ (left) and $\Phi = 15^\circ$ (right), considering a blade section at 0.25m distant from the tip in the less influenced zone

between the cases at $\Phi = 0^\circ$ and $\Phi = 15^\circ$ is equal to 1.8%, while downwind the blade experiences normal forces that increase with tilt angle.

Regarding the tangential force coefficient, which has a direct impact on the rotor torque and power, although slightly reduced values are experienced over the upwind cycle (maximum of 4.5% lower than the case at $\Phi = 0^\circ$), significant increases are experienced over the downwind cycle. Consequently this increase in the tangential force coefficient is the reason for the augmentation of the normalised power. In fact, figure 8 shows that the maximum value of c_n and C_t is reached for the case at $\Phi = 15^\circ$:

$$\frac{c_{n,MAX\{180^\circ < \theta < 360^\circ\}, \Phi=15^\circ}}{c_{n,MAX\{180^\circ < \theta < 360^\circ\}, \Phi=0^\circ}} = 2.32 \quad \frac{c_{t,MAX\{180^\circ < \theta < 360^\circ\}, \Phi=15^\circ}}{c_{t,MAX\{180^\circ < \theta < 360^\circ\}, \Phi=0^\circ}} = 7.1 \quad (4)$$

This beneficial effect is because the bottom part of the rotor in the downwind section is less influenced by the passage of the previous blade, i.e. less wake interference. This is shown in figure 10, showing colour shaded velocity contours on a plane parallel to the flow direction.

During the revolution, the wind over the upwind cycle is decelerated by the passage of the blades, i.e. there is a momentum loss associated with the work extracted by the rotor. This wake moves downstream in the wind direction creating a zone of low momentum flow that is crossed by the advancing blade. When the rotor is not inclined to the flow, the entire downwind blade encounters the wake from the upwind blade. However, at different

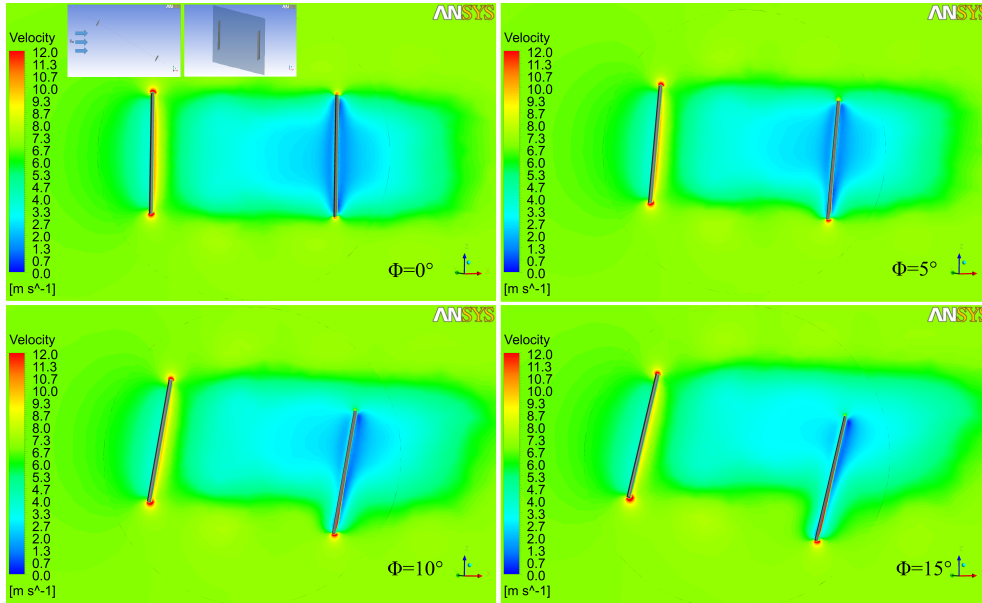


Figure 10: Velocity field at different tilt angle

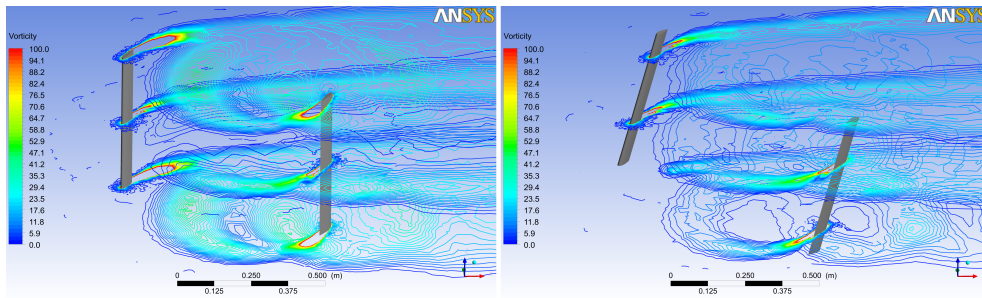


Figure 11: Vorticity contours at $\Phi = 0^\circ$ (left) and $\Phi = 15^\circ$ (right)

Φ a lower part of the blade is outside of this wake and thus experiences a higher dynamic pressure and an increase in local aerodynamic forces.

Similarly, figure 11 shows the effect of the wake from the passage of the preceding blade on the downwind cycle in terms of vorticity contours, and highlights that at higher tilt angles the vorticity is reduced in this region.

The static pressure field is shown in Figure 9 for a blade aerofoil section at $\theta = 240^\circ$ and at a distance of 0.25m from the lower blade tip (i.e. far enough to be not influenced by the tip vortices). This qualitative comparison shows some significant differences in the local pressure field as the rotor is tilted to $\Phi = 15^\circ$, especially on the suction surface, resulting in a higher value of lift and an augmentation in the power coefficient.

4. Conclusions

It has been shown that the CFD simulations using a URANS model are a promising way to investigate the complex phenomena that are present in an analysis of a VAWT. After initially validating the CFD model, the second part of the paper utilizes the method to consider the performance increments that have been observed in skewed flow. The 3D CFD model shows better prediction of the behaviour of the fluid, which involves unsteady three-dimensional effects such as dynamic stall. However, some differences from the experimental data have been displayed and it could be useful to investigate these further using more complex models, LES (Large Eddy Simulation) or DES (Detached Eddy Simulation) models for example, in order to give a more precise description of these dynamic phenomena. As regards the analysis of an H-VAWT in different tower tilting conditions, a quasi-static analysis has been performed, with the aim to separate and analyse the causes that lead to an increase of the coefficient of power c_p . The present work offers a possible interpretation of this phenomenon, which was studied through experimental tests and only briefly using a semi-empirical model by Mertens. This study finds that the gain is due to the downwind part of the rotor being less disturbed by the wake generated by the blades in the upwind part. Since the interest in the future use of VAWTs is primarily on floating platforms for offshore applications, it could be interesting to evaluate the influence of real motion on the rotor performances. However, it is important to note that it has been possible to determine and localize the physical mechanisms described above using a relatively simple numerical model, i.e. with only 2.5 millions of cells for a full scale turbine and about 2 million for the small turbine, a relatively simple turbulence model (two-equations model), and appropriate time steps that were found as a good compromise between computational costs and accuracy in results.

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Rebuttal to the peer review comments

3D URANS Analysis of a Vertical Axis Wind Turbine in Skewed Flows

Journal of Wind Engineering & Industrial Aerodynamics

Manuscript # INDAER-D-15-00226

Dear Editor/Reviewers,

We would like to thank you for consideration of our paper for publication in your prestigious journal. In particular, we would like to thank the peer reviewer since they allowed us to enhance the quality and the clarity of the paper taking into account their detailed comments. We have revised our manuscript, taking into account all the reviewers' comments.

We hope that the revised version of the manuscript will satisfy both the editor and the reviewers in order to proceed to publication; we look forward to hearing back from you and remain at your disposal for anything further.

Kind Regards,

All authors

Reviewer #1

1. *The mesh size used for each test case was presented. However, it is unclear whether the mesh size is sufficient to properly resolve the flow. A mesh refinement study should be presented as an evidence of adequate mesh resolution.*

Answer: The following explanation and figures have been added to the article.

For all the cases presented, both for the 2D and the 3D models, a mesh sensitivity analysis has been conducted, to assess the adequate mesh element number able to resolve the flow. As regards the 2D model, three meshes were analysed (coarse, medium, fine) as shown in the table below, keeping the wall distance fixed at $1.6 \cdot 10^{-5}$ with $y^+ < 1$.

Mesh	No. nodes on blade	No. elements	Wall distance
1	620	301163	$1.6 \cdot 10^{-5}$ chords
2	1240	464845	$1.6 \cdot 10^{-5}$ chords
3	1920	700292	$1.6 \cdot 10^{-5}$ chords

Table 1: Details of the 2D grids

The 3D mesh independence study was performed only for the skewed flow case. The same philosophy was then applied to the SANDIA case. For the 3D model of the turbine in skewed flow, two grids have been considered (medium and finer), as showed in Table 2. In general a larger number cases are analysed in order to conduct a sensitivity analysis, but in this case we considered the previous work done in this area (see cited articles). All the 3D meshes were built considering a wall distance equal to $1.9 \cdot 10^{-4}c$, resulting in $y^+ < 5$.

Mesh	Number of elements	Nodes around blades	Nodes in spanwise direction	Wall distance
4	1724930	84	200	$1 \cdot 10^{-4}$ chords
5	2672993	168	300	$1 \cdot 10^{-4}$ chords

Table 2: Details of the 3D grids for H-Darrieus turbine.

The results of these analyses are showed in Figure 1 and Figure 2, where the C_t (tangential force coefficient) values are compared.

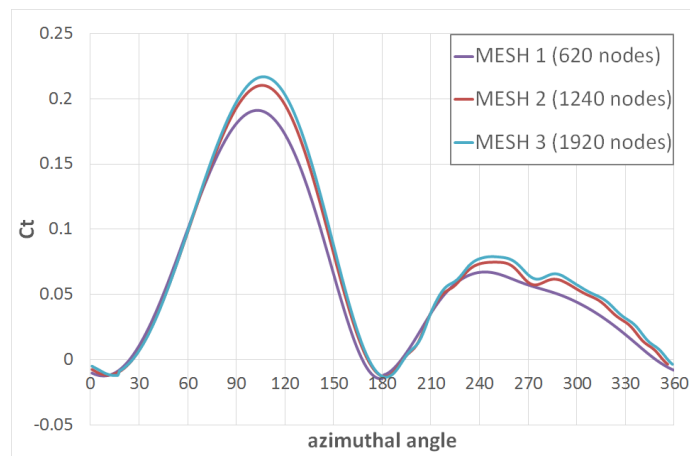


Figure 1: Mesh sensitivity analysis for the 2D model.

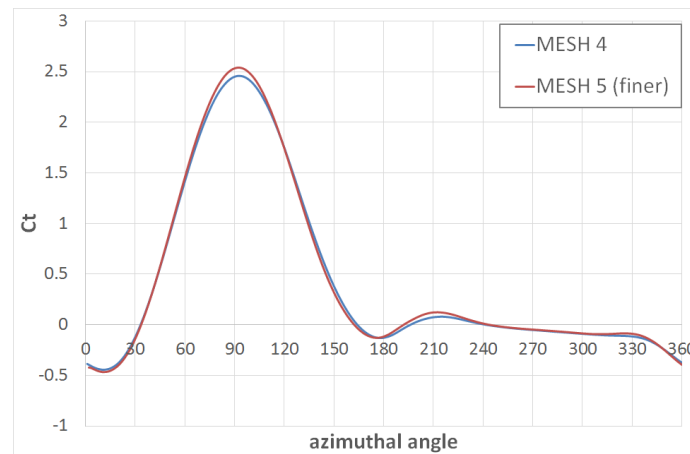


Figure 2: Mesh sensitivity analysis for the 3D model.

As regard the 2D analysis, the difference between mesh 2 and mesh 3 is minor, and mesh 2 was adopted, in order to save time and computational resources without substantially affecting accuracy. The same conclusion can be derived from the second graphs, showing a small gap between the two grids. So that “MESH 4” was used for the 3D analysis. Furthermore, this meshing philosophy was subsequently adopted for the SANDIA turbine case.

2. *Details on the CFD solver settings need to be included. What scheme and/or order were used for spatial and temporal discretization?*

Answer: The following text and table have been added to the article. The setup settings for the simulations are showed in Table 3:

Solver	
Type Time	<i>Pressure-Based Transient</i>
Solution Methods	
Pressure-Velocity Coupling	<i>PISO</i>
Spatial Discretization	
Gradient Pressure	<i>Green-Gauss Node Based Standard</i>
Momentum	<i>Second Order Upwind</i>
Turbulent Kinetic Energy	<i>Second Order Upwind</i>
Specific Dissipation Rate	<i>Second Order Upwind</i>
Modified Turbulent Viscosity	<i>Second Order Upwind</i>
Transient Formulation	
<i>Second Order Implicit</i>	

Table 3: Setup settings of CFD simulations.

The simulations are done using a First Order Implicit scheme for all the variables of the spatial discretization in the first revolutions, after that increase the order of the schemes considered to prevent the model from numerical diffusion errors. The convergence criteria were set at $1 \cdot 10^{-5}$ for all residuals.

3. *On page 4, it is unclear how a 2D simulation can be compared with a 3D simulation. Please clarify how force coefficients are calculated in 3D. Were they calculated at a specific airfoil section? Were the 3D force coefficients normalized by chord or area?*

Answer: All the experimental data refers to the equatorial plane of the turbine and measured 2D force coefficients are based on a surface integral of static pressure measurements as reported in Akins 1989. Due to the symmetry of the turbine it is expected that the flow in this plane is essentially 2-dimensional in nature, neglecting any wind shear effects, and should therefore compare favourably with a 2D CFD analysis.

4. *In order to properly study the influence of the wake, the simulation should run for at least 1 full rotation. Please list the time scale and time steps of the simulations.*

Answer: The following text has been added to the article after the first paragraph of section 3.1.

All results refers to the 7th revolution, where the difference between the average total torque coefficient C_m and the one of the previous revolution is within 1%.

SANDIA turbine has been analysed at two different wind conditions:

- TSR=4.6 (corresponding to a free stream velocity of 7.345 m/s) with a time scale corresponding to 1° /time-step (0.0043 s).
- TSR=2.33 (corresponding to a free stream velocity of 14.5 m/s) with different time scales corresponding to 1° /timestep (0.00431 s), 0.5° /time-step (0.00216 s) and 0.25° /time-step (0.00108 s).

As regards the simulations of the H-Darrieus turbine, the free-stream velocity was fixed at 7 m/s and all the results refer to a time scale equal to 1° /timestep at each TSR considered.

5. *Figures 2b and 2d showed a disturbingly large discrepancy between the experimental data and CFD results. Such discrepancy perhaps indicates issues with the CFD setup. Please address those discrepancies.*

Answer: The authors agree that the trends in Figures 2b and 2d seem to show a substantial difference between the experimental data and the CFD results. The following text has been added to the article in order to address this point.

As regard the discrepancies between the experimental and numerical prediction of C_t , the following should be considered. As very well known, since integrating pressure distributions to get the tangential force is much less accurate than integrating them to get the normal force, particularly if the pressure data is sparse, the accuracy of C_t is in general lower. Furthermore, it is shown that the CFD predicts well the trends at low speed, as expected when the flow is fully attached. Also, the power is well predicted at this condition, which depends on the tangential force and so suggests that the predicted tangential force is reasonably accurate. At the higher wind speeds the flow is much more complex, and so inaccuracies in predicting the dynamic flow are expected. Anyway, the overall power is still well predicted, which indicates again that tangential force estimation is reasonably accurate.

6. *Typo on page 4 line 42: where F_n and F_n are -> where F_n and F_t are*

Answer: Thanks. It has been corrected.

7. *On page 6, the authors attributed the difference between the simulation and experimental data to the over-production of wall shear stress. A further explanation is perhaps needed to clarify.*

Answer: As showed in Figure 3, k-e RNG model gives higher values of wall shear stresses than the one calculated by the k-w SST model. These generate forces which oppose the motion of revolution of the turbine, resulting a lower averaged torque coefficient and thus a lower power coefficient.

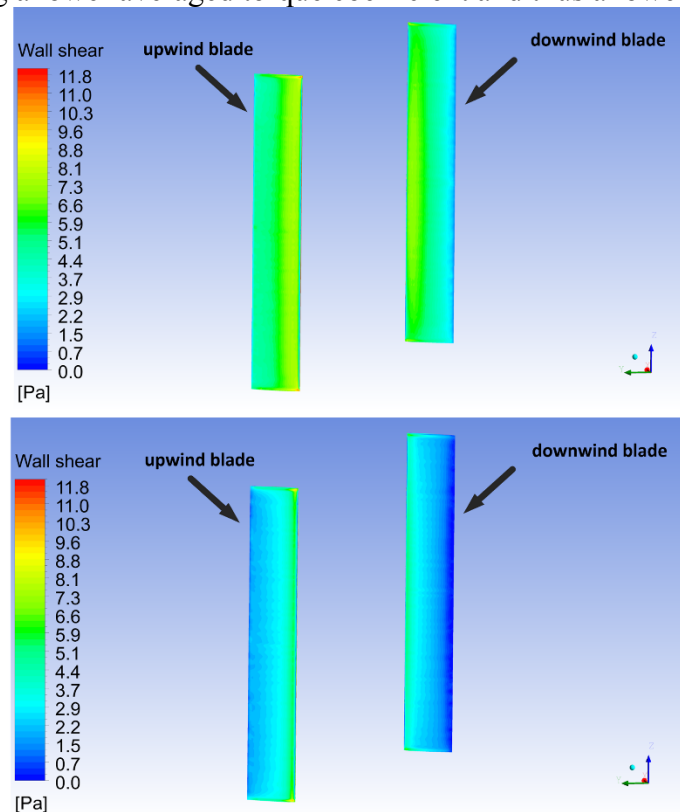


Figure 3: Wall shear on blades for k-e RNG model (top) and k-w SST model (bottom).

8. *Figure 5 seems to be one of the most important figures of the paper. The authors compared the C_p ratio at different tilt angles. What about the actual C_p instead of C_p ratio? It seems to be more important to compare the actual values instead of ratios.*

Answer: Unfortunately the absolute C_p values were not reported in the report considered, and only the the C_p ratios have been published.

A note clarifying this point has been added to the article: “The C_p ratio has been considered rather than the C_p since only the first was available in the report considered.”

Reviewer #2

1. *This study demonstrates the potential of an unsteady RANS 3D approach to predict the effects of skewed winds on the performance of an H-type vertical-axis wind turbine (VAWT). This paper is interesting.*

Answer: The authors would like to thank the second reviewer for the interest showed in the paper.