Using Nonlinear Normal Modes for Execution of Efficient Cyclic Motions in Soft Robots

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Abstract—With the aim of getting closer to the performance of the animal muscleskeletal system, elastic elements are purposefully introduced in the mechanical structure of soft robots. Indeed, previous works have extensively shown that elasticity can endow robots with the ability of performing tasks with increased efficiency, peak performances, and mechanical robustness. However, despite the many achievements, a general theory of efficient motions in soft robots is still lacking. Most of the literature focuses on specific examples, or imposes a prescribed behavior through dynamic cancellations, thus defeating the purpose of introducing elasticity in the first place.

This paper aims at making a step towards establishing such a general framework. To this end, we leverage on the theory of oscillations in nonlinear dynamical systems, and we take inspiration from state of the art theories about how the human central nervous system manages the muscleskeletal system. We propose to generate regular and efficient motions in soft robots by stabilizing sub-manifolds of the state space on which the system would naturally evolve. We select these sub-manifolds as the nonlinear continuation of linear eigenspaces, called nonlinear normal modes. In such a way, efficient oscillatory behaviors can be excited. We show the effectiveness of the methods in simulations on an elastic inverted pendulum, and experimentally on a segmented elastic leg.

Index Terms—Soft Robotics, Compliantly Actuated Robots, Robot Control, Nonlinear Normal Modes, Human Inspired Control.

I. INTRODUCTION

Actuation in living beings shows characteristics very different from the classic rigid robotic structures. Tendon and muscles elasticity, rather than being an impediment, enables animals to robustly interact with the external world and to efficiently perform dynamic and oscillatory tasks [1]. Inspired from the natural actuation, in soft robots elastic elements are purposefully introduced in the mechanical structure. Two main branches exist in soft robotic research according to their main source of inspiration. The first one involves robots made by continuously flexible materials. In analogy to the animal invertebrate body, the compliance is here distributed in the whole structure [2]. Notable examples are [3], [4], [5]. The other branch is instead inspired by the vertebrate muscleskeletal system. Here the compliance is mostly concentrated in the joints [6]. Such robots are typically referred as articulated soft robots or compliantly actuated robots. Examples are [7], [8], [9], [10], [11].

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Together with the new possibilities, soft robotics comes with the new challenge of developing control strategies being able to properly exploit the intelligence embedded in the robot mechanics [12], [13]. Several works exist in literature, where analytical optimal control is used to derive strategies fully exploiting the dynamics of soft robots, e.g. in terms of adaptability and safety [14], maximization of peak performances [15], efficient execution of cyclic motions [16]. However, while very effective in generating meaningful strategies, analytical optimal control is limited to low dimensional systems and specific tasks. Thus, to reach a full exploitation of the intrinsic soft robot dynamics, the investigation of more general paradigms is crucial.

In this work we specifically target the problem of generating regular oscillatory behaviors in soft robots with multiple degrees of freedom (see Fig. 1). Classical techniques of trajectory tracking can be used to reach this goal, such as feedback linearization [17], backstepping [18], adaptive control [19], just to cite a few. However, as discussed in [20], these approaches deeply change the plant inherent behavior. Instead of exploiting the intelligence embodied by design, they replace the natural dynamics with a different desired model, defeating the main purpose of introducing physical compliance. With the aim of overcoming this limitation novel approaches specifically designed for soft robotic systems were proposed. In [21] the model of an elastically actuated segmented leg is
matched to the SLIP one. In [22] joints dynamics is decoupled by control, canceling Coriolis, centrifugal and gravitational effects, in combination with energy regulation. In [23] virtual holonomic constraints are imposed, again in combination with energy regulation. However, these techniques still envisage a certain level of dynamics cancellation, which results only in a partial exploitation of the intrinsic system dynamics.

Looking instead at the natural world, humans are able to intuitively execute complex oscillatory movements, exploiting the musculoskeletal dynamics [24] despite the vast abundance of the body degrees of freedom. This ability was pinpointed for the first time by Nicolaj Bernstein [25], in the so-called motor equivalence problem. According to more recent neuro-scientific theories [26], [27], the central nervous system is able to implement such a behavior by stabilizing a set of variables of interest, while leaving the remaining degrees of freedom to evolve naturally. Regulating these variables implicitly identify a manifold, the so-called UnControlled Manifold, or UCM. The neural mechanism deputed to UCM stabilization is referred as synergy [28].

Inspired by these neuro-scientific evidences, we propose to exploit the soft robot’s embodied intelligence by stabilizing manifolds of reduced dimensionality on which the robot can naturally evolve. The evolution is natural in the sense that it is a direct expression of the autonomous dynamics of the physical system. The theory that studies these manifolds in dynamical system field is the so-called modal analysis, which is a classical result in linear systems theory. Nonlinear extensions of this concept are a more recent development, taking the name of Nonlinear Normal Modes [29]. To the best of authors knowledge, their application to robotic systems analysis and control is considered here for the first time, while the application of so-called similar normal modes were investigated in [30] (see Sec. III for more details).

In this work we consider the control of soft robots that can be modeled with a finite set of ordinary differential equations. This of course includes articulated soft robots, and we will focus mostly on them for the sake of tractability. However, in the past few years, several works [31], [32], [33], [34] demonstrated that continuously deformable robots can be described at any given level of accuracy through a finite dimensional discretization. So the proposed results are to be considered generally applicable also to continuous and hybrid kinds [35].

The work is organized as follow. In Sec. II the control strategy is introduced and discussed in the linear case. In Sec. III we survey the literature related to the non linear extensions of linear modal analysis. In Sec. IV the nonlinear case is faced, and a control algorithm assuring the modal manifold attractiveness is proposed. The problem of exciting a specific orbit on the manifold is also discussed. In Sec. V we apply the proposed strategy to the control of a spring loaded inverted pendulum. The modal analysis is discussed, and simulations are provided. Finally an experimental validation of the robustness of the control architecture is proposed in Sec. VI. Conclusions are drawn in Sec. VII.

II. LINEAR CASE

We consider as simplest prototype of soft robotic systems with \( n \) degrees of freedom (DoFs hereinafter), the following generic linear mechanical system

\[
M \ddot{x} = -K x + \tau, \tag{1}
\]

where \( x \in \mathbb{R}^n \) are the system’s configuration coordinates. \( K \in \mathbb{R}^{n \times n} \) such that \( K = K^T > 0 \) is the stiffness matrix. \( M \in \mathbb{R}^{n \times n} \) such that \( M = M^T > 0 \) is the inertia matrix. \( \tau \in \mathbb{R}^n \) are the generalized forces. For the sake of brevity, we consider the system to be conservative. However all the arguments that follow are easily generalizable to the dissipative case.

The natural oscillations of a linear mechanical system are well studied in the classic linear system theory. In this case unforced evolutions are always a linear combination of a finite set of normal modes, in number less or equal to the DoFs. The modal evolutions can be evaluated as complex exponential of the system’s eigenvalues. Each mode evolves in its own eigenspace spanned by the generalized eigenvectors associated to the mode. The projection of an eigenspace in the configuration space indicates the directions of oscillation. In non dissipative systems these directions can be directly evaluated through spectral decomposition of the matrix \( M^{-1}K \).

For a complete description of modal analysis and resonance in linear systems please refer to [36].

Thanks to these well known properties, designing control strategies exploiting the dynamics of a linear system to generate natural oscillatory behaviors is a relatively simple task. Let’s examine the problem of stabilizing the system in one of its eigenspaces. We consider without loss of generality the eigenspace spanned by the generalized eigenvectors associated to the first eigenvalue.

**Proposition 1.** The system (1) can be stabilized along the eigenspace associated to its first eigenvalue \( \lambda_1 \), through the feedback control action

\[
\tau = -\beta M \begin{bmatrix} 0_{m_1 \times m_1} & 0_{m_1 \times n-m_1} \\ 0_{n-m_1 \times m_1} & I_{n-m_1 \times n-m_1} \end{bmatrix} V^T \dot{x}. \tag{2}
\]

Each column of \( V \in \mathbb{R}^{n \times n} \) is a distinct generalized eigenvector of the matrix \( M^{-1}K \) in (1), ordered such that the first \( m_1 \) left columns are a base of the eigenspace associated to \( \lambda_1 \).
III. NONLINEAR NORMAL MODES

A. Background

Let us consider a generic nonlinear mechanical system with \( n \) degrees of freedom in the form

\[
\ddot{x}_j = f_j(x, \dot{x}),
\]

where \( x = [x_1, \ldots, x_n] \in \mathbb{R}^n \) are the system configuration coordinates, \( \dot{x} = [\dot{x}_1, \ldots, \dot{x}_n] \in \mathbb{R}^n \) their derivatives, and \( f_j(x, \dot{x}) : \mathbb{R}^{2n} \rightarrow \mathbb{R} \) is the dynamics of the \( j \)-th DoF.

Generalizing the powerful linear modal analysis to nonlinear mechanical systems occupies the dynamical system theorists since Lyapunov times (see e.g. [37]). In his seminal work [38], Rosenberg defines a nonlinear normal mode as "vibrations in unison" where all the material points of the system reach their extremal points and cross the origin simultaneously. The modes are called similar or rectilinear if they move in a flat space, and non-similar otherwise. While similar normal modes are not typical in nonlinear systems, nontrivial examples exist in literature [39], [30]. Non-similar Normal Modes are well studied, e.g. from the point of view of resonances [40], localization [41], and bifurcations [42]. However Rosenberg's definition has two major limitations: i) in resonant conditions the relative phases of the oscillations can change, violating the unison hypothesis; ii) it requires the system to be conservative.

In their 1993 work [43], Shaw and Pierre proposed an alternative extension of linear modes overcoming these limitations, by generalizing eigenspaces to curvilinear spaces through the concept of invariant manifolds. A manifold is invariant w.r.t a dynamics, if the vector field describing the system dynamics is always tangent to the manifold, i.e. if a trajectory initialized on the manifold remains on it. As discussed in the previous section, this invariance property is a main characteristic of eigenspaces in linear system, which are then invariant manifold.

Shaw-Pierre Nonlinear Normal Modes (NNM) were defined as "a motion which takes place on a two-dimensional invariant
manifold in the system’s phase space”, which is tangent to an eigenspace of the linearized system in an equilibrium point. We will refer in the following to such invariant manifold, as modal manifold.

For the sake of brevity and clarity, we introduce the following common assumptions

i We consider the modal manifold to be parameterizable through two independent variables [44], which are typically selected as one configuration coordinate and its time derivative [43].

ii We select $x_1, \dot{x}_1$ as independent variables.

iii We assume the system equilibrium to be such that $x_1 = 0$.

Note that the first assumption constraints the NNM to be a continuation of a linear mode unidimensional in configuration space. In other words, the algebraic multiplicity $m_1$ of the first eigenvalues $\lambda_1$ of the linearized system is equal to 1. Second and third assumptions are instead imposed without loss of generality. It is convenient, but not necessary, to perform a linear change of variables such that $x_1$ points in the direction of modal oscillation, as e.g. in (54).

Under these three assumptions the manifold can be implicitly defined by the set of nonlinear algebraic equations

$$x_j = X_j(x_1, \dot{x}_1), \quad \dot{x}_j = \dot{X}_j(x_1, \dot{x}_1) \quad \forall j \in \{1 \ldots n\},$$  

(7)

where $X_j : \mathbb{R}^2 \rightarrow \mathbb{R}$ and $\dot{X}_j : \mathbb{R}^2 \rightarrow \mathbb{R} \forall j \in \{1 \ldots n\}$ (hereinafter also called maps). For $j = 1$ the maps have the trivial forms $X_1(x_1, \dot{x}_1) = x_1$ and $\dot{X}_1(x_1, \dot{x}_1) = \dot{x}_1$. Using (7), the dynamics of $x_1$ on the manifold can be expressed independently from the values of $x_2, \ldots, x_n$ as

$$\begin{align*}
\dot{x}_1 &= F_m(x_1, \dot{x}_1) \\
F_m(x_1, \dot{x}_1) &\triangleq f_1(X(x_1, \dot{x}_1), \dot{X}(x_1, \dot{x}_1)),
\end{align*}$$  

(8)

where $X(x_1, \dot{x}_1) : \mathbb{R}^2 \rightarrow \mathbb{R}^{2n}$ and $\dot{X}(x_1, \dot{x}_1) : \mathbb{R}^2 \rightarrow \mathbb{R}^{2n}$ are vector valued functions having as $j$-th element $X_j$ and $\dot{X}_j$ respectively, and $F_m(\cdot, \cdot)$ specifies the modal dynamics.

Thus, if initialized on the modal manifold the mechanical system is equivalent to the one dimensional dynamics (8), which drives the remaining $n-1$ degrees of freedom through the set of algebraic relationships (7). This resembles the behavior of a linear system initialized on one of its eigenspaces. We summarize such physical interpretation in Fig. 4. To further underline this distinction, in literature $x_1$ is referred as master variable, and $x_2, \ldots, x_n$ as slave variables.

B. Evaluation of the Invariant Manifold

The manifold geometry can be connected to the system dynamics by deriving (7), and substituting the vector field (6) on the manifold. This yields to the set of tangency constraints

$$\begin{align*}
\dot{X}_j &= \frac{\partial X_j}{\partial x_1} \dot{x}_1 + \frac{\partial X_j}{\partial \dot{x}_1} \ddot{x}_1 \\
&\quad + \frac{\partial X_j}{\partial \dot{x}_1} \dot{f}_1(X, \dot{X}) \\
f_j(X, \dot{X}) &= \frac{\partial X_j}{\partial x_1} \dot{x}_1 + \frac{\partial X_j}{\partial \dot{x}_1} \ddot{x}_1 + \frac{\partial X_j}{\partial \dot{x}_1} \dot{f}_1(X, \dot{X}) \\
&\quad \forall j \in \{2 \ldots n\},
\end{align*}$$  

(9)

Note that the dot on top of $X$ should not be considered as a time derivation. This abuse of notation is instrumental to simplify the notation afterwards.

The basic idea is to apply some constraints on the functional spaces in which $X_j, \dot{X}_j, f_j$ live, to express them with a finite set of basis functions. This allows to approximate the PDEs with a set of algebraic equations. Approaches of this type that were employed so far are Taylor expansion [43], Koopman operator [46], and Harmonic balance [47]. In alternative to this approach, several numerical methods were proposed in literature, as e.g. finite element analyses. See [48] for an extensive review. The result is a numerical approximation of the PDE solution. For further details on theoretical and applicative results in Nonlinear Normal Modes please refer to [29] and [49] respectively.
IV. Controlling Nonlinear Normal Modes

As discussed in Sec. II for the linear case, we propose here to generate efficient nonlinear oscillations in soft robots by stabilizing a modal invariant manifold, which thus assumes the role of an artificial counterpart of the natural UCM [26]. We will also briefly discuss the problem of injecting or removing energy in the system, in order increase or decrease the amplitude of modal oscillations.

A. General model definition

Accordingly to the standard formulation in [50], a generic soft robot can be modeled as

\[
\begin{bmatrix}
M(x_m) & Q(x_m) \\
Q^T(x_m) & B(x_m)
\end{bmatrix}
\begin{bmatrix}
\dot{x}_i \\
\dot{x}_m
\end{bmatrix} + c(x_m, \dot{x}_m, \dot{x}_i) + g(x_m, \dot{x}_i)
\]

\[
+ \frac{\partial V(x_m)}{\partial x_m} + d(x_m) = \begin{bmatrix} \tau_i \\ \tau_m \end{bmatrix} \tag{10}
\]

where \(x_m\) are the configuration coordinates associated to the motors, and \(x_i\) is the robot configuration. \(M(x_m, \dot{x}_m), B(x_m)\) and \(Q(x_m, \dot{x}_m)\) are inertia matrices of links, motors and coupling respectively. \(c(x_m, \dot{x}_m, \dot{x}_i)\) collects Coriolis and centrifugal effects, \(g(x_m, \dot{x}_i)\) models the gravity torque, and \(V(x_m)\) is the elastic potential. \(\tau_i\) and \(\tau_m\) are the actuations. Note that, while formulated with articulated soft robots in mind, this definition also includes discrete models of continuous soft robots [32], [34].

By considering \(x = [x_i^T, x_m^T]^T\),

\[
\tau = \begin{bmatrix} M(x_i, x_m) & Q(x_i, x_m) \\
Q^T(x_i, x_m) & B(x_i, x_m) \end{bmatrix}^{-1} \begin{bmatrix} \tau_i \\ \tau_m \end{bmatrix}, \tag{11}
\]

and

\[
f(x, \dot{x}) = -\begin{bmatrix} M(x_i, x_m) & Q(x_i, x_m) \\
Q^T(x_i, x_m) & B(x_i, x_m) \end{bmatrix}^{-1} \begin{bmatrix} c(x_i, x_m, \dot{x}_m, \dot{x}_i) \\ g(x_m, \dot{x}_i) + \frac{\partial V(x_m)}{\partial x_m} + d(x_m) \end{bmatrix}, \tag{12}
\]

Eq. (10) is rewritten as the actuated version of the generic nonlinear mechanical system (6). Thus the dynamical model of the \(j\)-th configuration coordinates of a soft robot that we consider in the following is

\[
\dot{x}_j = f_j(x, \dot{x}) + \tau_j, \tag{13}
\]

where \(\ddot{x}_j, \dot{x}_j, x_j, \tau_j, \) and \(f_j\) are the \(j\)-th elements of \(\ddot{x}, \dot{x}, x, \tau, \) and \(f\) respectively.

We introduce \(\Sigma: \mathbb{R}^2 \to \mathbb{R}^{n-1 \times n-1}\) and \(\Gamma: \mathbb{R}^2 \to \mathbb{R}^{n-1 \times n-1}\), with elements \((i, j)\) defined as

\[
\Sigma_{i,j}(x_1, \dot{x}_1) = \frac{\partial f_{i+1}}{\partial x_{j+1}} \bigg|_{x=x, \dot{x} = \dot{x}} \tag{14}
\]

\[
\Gamma_{i,j}(x_1, \dot{x}_1) = \frac{\partial f_{i+1}}{\partial x_{j+1}} \bigg|_{x=x, \dot{x} = \dot{x}}.
\]

These two matrix-valued functions generalize the role that damping and stiffness had in the linear case. Note indeed that \(\Sigma(0, 0)\) is the stiffness of the slave variables in the equilibrium point, and \(\Gamma(0, 0)\) the damping.

We refer the total energy of the soft robot (13) as \(E(x, \dot{x})\). We also define the equivalent energy on the manifold as

\[
E_M(x_1, \dot{x}_1) = E(X(x_1, \dot{x}_1), \dot{X}(x_1, \dot{x}_1)). \tag{15}
\]

B. Stabilization

In this work we consider the following definition of manifold stability

**Definition 1.** Given a manifold parametrized by the two maps \(X(x_1, \dot{x}_1)\) and \(\dot{X}(x_1, \dot{x}_1)\) in (7), we call it locally attractive if there exists such that \(\|x_1(0) - X(x_1(0), \dot{x}_1(0))\| + \|\dot{x}_1(0) - \dot{X}(x_1(0), \dot{x}_1(0))\| < \delta\) \(\forall i\in \{1 \ldots n\}, \) then it holds

\[
\lim_{t \to \infty} x_i(t) - X(x_1(t), \dot{x}_1(t)) = 0 \quad \forall i \in \{1, \ldots, n\}. \tag{16}
\]

The following theorem generalizes Proposition 1 to the nonlinear case, providing a control strategy able to regulate the soft robot on one of its modal manifolds.

**Theorem 1.** Let \(X(x_1, \dot{x}_1)\) and \(\dot{X}(x_1, \dot{x}_1)\) be the parametrization of a nonlinear modal manifold for the \(n\)-DoF nonlinear mechanical system (6).

Then the feedback law

\[
\tau(x, \dot{x}) = \begin{bmatrix} 0 & 0 \\
0 & I_{n-1} \end{bmatrix} (\kappa_p \Delta + \kappa_d \dot{\Delta}) + \begin{bmatrix} \tau_1(x, \dot{x}) \\
\vdots \\
0 \end{bmatrix}, \tag{17}
\]

where \(\kappa_p \in \mathbb{R}^+, \kappa_d \in \mathbb{R}^+, \Delta = X(x_1, \dot{x}_1) - x, \) and

\[
\tau_1(x, \dot{x}) = f_1(X(x_1, \dot{x}_1), \dot{X}(x_1, \dot{x}_1)) - f_1(x, \dot{x}), \tag{18}
\]

preserves the invariance of the manifold.

Furthermore, the two following sufficient conditions for the local attractiveness hold

i) \(\delta \in \mathbb{R}^+\) always exists such that if

\[
\|\Sigma(x_1, \dot{x}_1)\| < \delta, \quad \|\Gamma(x_1, \dot{x}_1)\| < \delta
\]

then the manifold is attractive \(\forall \kappa_p, \kappa_d\) such that

\[
\lambda^- (\Sigma(x_1, \dot{x}_1)) > -\kappa_p, \quad \lambda^- (\Gamma(x_1, \dot{x}_1)) > -\kappa_p,
\]

where \(\lambda^- (\cdot)\) is the minimum eigenvalue of the argument.

ii) If \(\Sigma(x_1, \dot{x}_1)\) and \(\Gamma(x_1, \dot{x}_1)\) are simultaneously diagonalizable by a matrix \(\mu(x_1, \dot{x}_1) \in \mathbb{R}^{n-1 \times n-1}\), then the manifold is attractive if

\[
\lambda_i(\Sigma(x_1, \dot{x}_1)) > -\kappa_p, \quad \lambda_i(\Gamma(x_1, \dot{x}_1)) > -\kappa_d - \lambda_i(\Sigma(x_1, \dot{x}_1)) - \kappa_p \tag{19}
\]

where \(\lambda_i(\cdot)\) extracts the eigenvalue corresponding to the \(i\)-th column of \(\mu(x_1, \dot{x}_1)\).

**Proof.** On the manifold (i.e. when \(x_i = X_i(x_1, \dot{x}_1)\) and \(\dot{x}_i = \dot{X}_i(x_1, \dot{x}_1)\)) \(\tau_i \equiv 0\) and \(\tau_j \equiv 0\). Thus, condition (9) is identical for the actuated and not actuated system, which proves the first thesis.
To prove the attractiveness, we define the displacement w.r.t.
the manifold as (see Fig. 5)
\[ \Delta_j = X_j(x_1, \dot{x}_1) - x_j, \]  
(20)

Deriving w.r.t. time yields
\[ \dot{\Delta}_j = -\dot{x}_j + \frac{\partial X_j}{\partial x_1} \dot{x}_1 + \frac{\partial X_j}{\partial \dot{x}_1} [f_1(x, \dot{x}) + \tau_1] \]
(21)

where in the first step we used the chain rule, and in the second
step we used (9), i.e. the manifold invariance. Now, we close
the loop with (18) obtaining
\[ \dot{\Delta}_j = \dot{X}_j(x_1, \dot{x}_1) - \dot{x}_j, \]
(22)

which now describes the displacement between velocities and
their corresponding manifold coordinate. We derive (22) a second
time obtaining
\[ \ddot{\Delta}_j = -f_1(x, \dot{x}) - \tau_j + \frac{\partial X_j}{\partial x_1} \ddot{x}_1 + \frac{\partial X_j}{\partial \dot{x}_1} [f_1(x, \dot{x}) + \tau_1] \]
(23)

Taking again the control (18) into account. Now, by exploiting
the manifold invariance and substituting (20) and (22), we write
\[ \ddot{\Delta}_j = f_1(X(x_1, \dot{x}_1), \dot{X}(x_1, \dot{x}_1)) \]
(24)

To complete the proof we linearize the dynamics around
a generic equilibrium trajectory on the manifold, i.e. for \( \Delta_j = 0 \)
and \( \dot{\Delta}_j = 0 \) \( \forall j \in \{2 \ldots n\} \)

\[ \begin{bmatrix}
\xi \\
\sigma
\end{bmatrix} \simeq \begin{bmatrix}
0 \\
-(\Sigma(x_1, \dot{x}_1) + \kappa_0 I) & I
\end{bmatrix} \begin{bmatrix}
\xi
\
sigma
\end{bmatrix}, \]
(25)

where \( \xi = [\Delta_2 \ldots \Delta_n]^T \) and \( \sigma = [\Delta_2 \ldots \Delta_n]^T \), where we
exploited that \( \Delta_1 \equiv 0 \) and \( \dot{\Delta}_j \equiv 0 \) by construction. In (25), \( x_1 \)
and \( \dot{x}_1 \) do not appear as an input, but only as dependencies
in the dynamic matrix. Indeed it holds
\[ \frac{\partial f_2(x_1, \dot{x}_1, X_2 \ldots X_n)}{\partial x_1} = \frac{\partial f_2(x_1, \dot{x}_1, X_2 \ldots X_n - \Delta_2 \ldots X_n)}{\partial x_1} \]
(26)

\[ \frac{\partial f_2(x_1, \dot{x}_1, X_2 \ldots X_n)}{\partial \dot{x}_1} = \frac{\partial f_2(x_1, \dot{x}_1, X_2 \ldots X_n - \Delta_2 \ldots X_n)}{\partial \dot{x}_1} \]
(27)

Note that the controller \( \tau_j \) decouples the dynamics of master variable from the slave variables. Indeed it holds
\( \dot{x}_1 = f_1(x, \dot{x}) + (f_1(X(x_1, \dot{x}), \dot{X}(x_1, \dot{x}_{1}))) \)
and \( f_1(X(x_1, \dot{x}), \dot{X}(x_1, \dot{x}_{1})) = f_1(X(x_1, \dot{x}), \dot{X}(x_1, \dot{x}_{1})) \), i.e. \( x_1 \)
evolves accordingly to the modal dynamics (8) also outside the manifold. Thus, the dependency of Eq. (25) from \( x_1 \) and \( \dot{x}_1 \) can be regarded as a time-variace, and (i) is directly proven by applying the
Lemma 1 in appendix B.

For proving (ii), we consider that two matrices are simul-taneous diagonalizable if and only if they commute [51]. Hence,
the hypothesis of simultaneous diagonalizability of \( \Sigma \) and \( \Gamma \)
implies that they commute, which in turn assures that also
\( \Sigma(x_1, \dot{x}_1) + \kappa_0 I \) and \( \Gamma(x_1, \dot{x}_1) + \kappa_3 I \) commute
\[ (\Sigma(x_1, \dot{x}_1) + \kappa_0 I)(\Gamma(x_1, \dot{x}_1) + \kappa_3 I) \]
(28)

Thus the thesis results from the application of Lemma 2 in appendix B, considering the hypothesis (19).

The proposed control law (17) is indeed a nonlinear general-
ization of (2):

a) If \( \Sigma \prec 0 \) the damping injection is sufficient to stabilize
the manifold, i.e. \( \kappa_0 = 0 \), as in the linear case.

b) We need here the extra control action \( \tau_j = f_1(X, \dot{X}) - f_1(x, \dot{x}), \)
which is essentially non-local. Indeed, due to the tangency property of the modal invariant manifold,
in the case of \( x_1 \) pointing in the direction of the first
eigenvector of the linearized system, \( \frac{\partial \Sigma}{\partial x_1} \bigg|_{x_1=0, \dot{x}_1=0} = 0 \)
and \( \frac{\partial \Gamma}{\partial x_1} \bigg|_{x_1=0, \dot{x}_1=0} = 0 \).

c) Note also that we are expressing the control in modal
coordinates (i.e. \( x_1 \) master variable) for the sake of clarity
and conciseness. Thus no change of coordinates (cf. \( V \) in
(2)) explicitly appears in (17). It should also be noticed that
the pre-multiplications for the inertia matrix present in
(2) are happening implicitly in the non linear case through
(11) and (12).

C. Orbit excitation
In this section we investigate the possibility of injecting or
removing energy in order to increase or decrease the amplitude
of oscillation. To this end, we start by asking under which
conditions it is possible to generate a control action \( \tau(x, \dot{x}), \)
that does not vanish on the manifold (i.e. \( \tau(X, \dot{X}) \neq 0 \)) and
such that the closed loop manifold is parametrized by the same
maps \( X \) and \( X \) of the open loop one. This is equivalent to impose that both (6) and (13) verify (9) for a same \( X \) and \( X \).

So, let's start by considering the second set of equations in (9),
\[ \begin{cases}
\frac{\partial f_2(x, \dot{x})}{\partial x_1} = \frac{\partial f_2(x_1, \dot{x}_1, X_2 \ldots X_n)}{\partial x_1} + \frac{\partial f_2(x_1, \dot{x}_1, X_2 \ldots X_n)}{\partial \dot{x}_1} \frac{\partial X_1}{\partial x_1} + \frac{\partial X_1}{\partial \dot{x}_1}, \\
\frac{\partial f_2(x, \dot{x})}{\partial \dot{x}_1} = \frac{\partial f_2(x_1, \dot{x}_1, X_2 \ldots X_n)}{\partial \dot{x}_1} + \frac{\partial f_2(x_1, \dot{x}_1, X_2 \ldots X_n)}{\partial \dot{x}_1} \frac{\partial X_1}{\partial x_1} + \frac{\partial X_1}{\partial \dot{x}_1}
\end{cases} \]
(29)

where \( \tau_j \) is the \( j-th \) element of \( \tau \). By subtracting the first
equation from the second, the following condition results
\[ \tau_j = \frac{\partial X_j}{\partial x_1} \frac{\partial \dot{x}_j}{\partial x_1}, \]
(30)

which prescribes how to exert \( \tau_j \), given \( \tau_j \).

Let's consider now the first equation in (9), for the two
systems (6) and (13)
\[ \begin{cases}
\frac{\partial X_j}{\partial x_1} \frac{\partial \dot{x}_j}{\partial x_1} = \frac{\partial f_2(x_1, \dot{x}_1, X_2 \ldots X_n)}{\partial \dot{x}_1} + \frac{\partial f_2(x_1, \dot{x}_1, X_2 \ldots X_n)}{\partial \dot{x}_1} \frac{\partial X_1}{\partial x_1} + \frac{\partial X_1}{\partial \dot{x}_1} \\
\frac{\partial X_j}{\partial x_1} = \frac{\partial f_2(x_1, \dot{x}_1, X_2 \ldots X_n)}{\partial \dot{x}_1} + \frac{\partial f_2(x_1, \dot{x}_1, X_2 \ldots X_n)}{\partial \dot{x}_1} \frac{\partial X_1}{\partial x_1} + \frac{\partial X_1}{\partial \dot{x}_1}
\end{cases} \]
(31)
Subtracting the two yields
\[ \frac{\partial X_j}{\partial \dot{x}_j} \tau_j = 0 \quad \forall j. \quad (32) \]
This equation leads to two possible scenarios. If
\[ \frac{\partial X_j}{\partial \dot{x}_j} = 0, \quad (33) \]
then \( \tau_j \) can be freely exerted. Note that (33) means \( x_j = X_j(x_1) \), i.e. the manifold is described by a set of virtual holonomic constraints. In this case exerting \( \tau \) such that (30) holds would be sufficient to inject energy into the system without ruining the invariance of the modal manifold.

However, this circumstance is rather restrictive. For this reason we will not consider it further in the paper. Under this assumption, (32) leads to \( \tau_j \equiv 0 \), which in turn leads to \( \tau_j \equiv 0 \) \( \forall j \in \{2 \ldots n\} \) through (30). Thus exerting any generalized force always pushes the system away from the manifold.

Adopting the philosophy of [52], the following theorem proposes a simple extension of the stabilizing controller of Theorem 1, which is able to regulate the system energy within a certain interval despite the discussed limitations.

**Theorem 2.** Let the soft robot (13), with a modal manifold parametrized by \( X(x_1, \bar{x}_1) \) and \( \dot{X}(x_1, \bar{x}_1) \), controlled through the feedback law
\[ \tau(x, \dot{x}) = \begin{bmatrix} 0 & 0 \\ 0 & I_{n-1} \end{bmatrix} \left( \kappa_D \dot{x} + \kappa_A \dot{\bar{x}} \right) + \begin{bmatrix} \tau_1(x, \dot{x}) + \tau_2(x_1, \bar{x}_1) \\ 0 \\ \vdots \\ 0 \end{bmatrix}, \]
with \( \tau_1 \) as in (18), and
\[ \tau_2 = \gamma \begin{cases} 0 & \text{if } x_1 \notin [x_1^-, x_1^+] \cup E_M \in [E^-, E+] \\ 1 & \text{if } x_1 \in [x_1^-, x_1^+] \\wedge \left((E_M < E^- \wedge \dot{x}_1 > 0) \lor (E_M > E^+ \wedge \dot{x}_1 < 0)\right) \\ -1 & \text{otherwise} \end{cases} \]
(34)
where \( \gamma > 0 \), \( E^+ > E^- > 0 \), and \( x_1^+ > 0 > x_1^- \) are scalar constants. If the following conditions hold simultaneously
H1 the soft robot is conservative, i.e. \( \frac{dE_M}{dr} \bigg|_{\tau_1=0} = 0 \)
H2 the level curves of \( E_M \) are closed
H3 \( f_1(X(x_1, 0), \dot{X}(x_1, 0)) \neq 0 \quad \forall x_1 \notin [x_1^-, x_1^+] \) and \( f_1(X(x_1, 0), \dot{X}(x_1, 0)) \neq \gamma \quad \forall x_1 \in [x_1^-, x_1^+] \)
H4 the hypotheses of Theorem 1 are verified
then the invariant manifold is locally attractive and
\[ \lim_{t \to \infty} E(x(t)) \in [E^-, E^+]. \]

**Proof.** We consider the case \( E(x_1(0), \dot{x}_1(0)) < E^- \), also sketched in Fig. 6. The case \( E(x_1(0), \dot{x}_1(0)) > E^+ \) can be derived with similar arguments. For this initial condition the closed loop dynamics of the master variable is
\[ x_1 = f_1(X(x_1, \bar{x}_1), \dot{X}(x_1, \dot{x}_1)) \]
\[ = \begin{cases} 0 & \text{if } x_1 \notin [x_1^-, x_1^+] \\ \gamma & \text{if } x_1 \in [x_1^-, x_1^+] \wedge \dot{x}_1 > 0 \\ \gamma & \text{otherwise} \end{cases} \]
(36)
which is an autonomous dynamics, not depending on the evolution of slave variables \( x_2, \ldots, x_n \). As common in the theory of hybrid systems [53], we make a partition of the time into a sequence of intervals
\[ [0, t) = \left( \bigcup_{i=1}^{i_{in}} t_i^+ \right) \cup \left( \bigcup_{j=1}^{j_{out}} t_j^- \right) \cup \left( \bigcup_{k=1}^{k_{in}} t_k^0 \right), \]
(37)
where \( t_i^+ \) is the \( i \)-th interval for which \( x_1 \in [x_1^-, x_1^+] \) and \( \dot{x}_1 \neq 0 \), \( t_j^- \) is the \( j \)-th interval for which \( x_1 \notin [x_1^-, x_1^+] \), and \( t_k^0 \) is the \( k \)-th interval for which \( x_1 \in [x_1^-, x_1^+] \) and \( \dot{x}_1 = 0 \). Thus \( t_i^+ \), \( t_j^- \), and \( t_k^0 \) are the number of intervals \( t_i^+ \), \( t_j^- \), and \( t_k^0 \) contained in \( [0, t) \).

Conditions \( x_1 \in [x_1^-, x_1^+] \) and \( \dot{x}_1 = 0 \) hold only for isolated instants, since H3 leads to \( \dot{x}_1 \neq 0 \). Thus \( t_i^+ \) are all of zero measure.

If \( x_1(t) \notin [x_1^-, x_1^+] \) then (36) becomes
\[ \dot{x}_1 = f_1(X(x_1, \bar{x}_1), \dot{X}(x_1, \dot{x}_1)), \]
which is the master variable’s dynamics on the manifold (8). Thus H1 leads to
\[ \frac{dE_M}{dr}(x_1, \dot{x}_1) = \dot{x}_1 \begin{cases} +\gamma & \text{if } \dot{x}_1 > 0 \\ -\gamma & \text{otherwise} \end{cases}, \]
(39)
Thus
\[ E_M(t) = E_M(0) + \int_0^t \frac{dE_M}{dr} dr' \]
\[ = E_M(0) + \int_0^t \frac{dE_M}{dr} dr'' \]
\[ \geq E_M(0) + \varepsilon \bar{i} \]
(40)
where \( \bar{i} = \sum_{i} (t_i^+_{in}) \max(t_{in}^+), \) and in the second step we changed the integral coordinate to express the time as union of \( t_{in}^+ \) intervals.

Eqs. (39) and (40) imply that \( E_M(t) \) is increasing for \( E_M < E^- \). Thus \( x_1 \) and \( \bar{x}_1 \) eventually reach a value such that \( E_M = E^- \). H1 assures that the system in open loop is conservative, thus once reached this condition \( E_M \) remains in \( [E^-, E^+] \).

Therefore a \( t \in \mathbb{R} \) always exists such that \( E_M(x_1(t), \bar{x}_1(t)) \in [E^-, E^+] \) for all the \( t > T \). This implies that \( \bar{x}_1(t) = 0 \), and (34) is equal to (17) for all the \( t > T \). Invoking Theorem 1, the manifold attractiveness follows from H4, and in turn
\[ \lim_{t \to \infty} E(x(t)) = \lim_{t \to \infty} E_M(x_1(t), \bar{x}_1(t)) = E^- \in [E^-, E^+]. \]

**Remark 1.** It is worth mentioning that both controllers (17) and (34) are such that \( \tau_i \to 0, \) since
V. THE GENERALIZED SLIP CASE OF STUDY

In the following we present the application of the proposed strategy to a simple yet relevant example of soft robot: the radial elastic pendulum. This system generalize the so-called spring-loaded inverted pendulum (SLIP) by the introduction of a polar spring. From the bio-mechanical and robotic point of view, SLIP models assume a wide interest if considered that the center of mass evolution for legged animals can be generated as a trajectory of the SLIP model, both during running and walking gates [24], [54]. Existing control strategies implement locomotion by matching the robot dynamics to the SLIP one [21], [55].

A. Dynamical Model

The elastic pendulum model is

\[
\begin{align*}
\dot{\theta} &= -2\frac{\kappa_1}{r} \dot{\theta} + \frac{\kappa_1}{r} \sin(\theta) - \frac{\kappa_2}{r^2} \dot{\theta} \\
\dot{\theta}_\theta &= +r \dot{\theta}^2 - g \cos(\theta) - \kappa_2 (r - r_0)
\end{align*}
\]

(41)

where \(\theta\) and \(r\) are the polar coordinates of the body, with their derivatives \(\dot{\theta}, \dot{\theta}_\theta, \dot{r}, \dot{r}_\theta, g\) is the gravity constant. \(\kappa_1\) and \(\kappa_2\) are the ratio between spring stiffness and the body mass, i.e. what in the linear case would have been called natural frequencies of oscillation.

The system has an equilibrium in \(\theta = 0\) and \(r = r_0 - \frac{g}{\kappa_2}\). Its linearized dynamics is \(\dot{\theta} \approx -\frac{\kappa_1}{(r_0 - \frac{g}{\kappa_2})^2} \theta\), and \(\dot{\theta}_\theta \approx -\kappa_2 \Delta r\). So, the linearized system normal modes are two decoupled evolutions along each degree of freedom: an angular oscillation with fixed radius, and a radial oscillation with fixed angle. The nonlinear extension of the latter radial mode remains a linear oscillation in the direction of the radial degree of freedom, since for \(\theta \approx 0\) and \(\dot{\theta} \approx 0\) the dynamics collapses into a linear one. The other mode instead turns into a much more complex oscillation which we investigate in the next subsections.

B. Analytical Approximated Solution of the Complete System

By considering as master variables \(\theta, \dot{\theta}, \dot{r}, \dot{r}_\theta\), (9) and (41) yield

\[
\begin{align*}
\frac{\partial X}{\partial \theta} + \frac{\partial X}{\partial \dot{\theta}} (-2\dot{\theta}) + \frac{\partial X}{\partial X} (-2\dot{X}) + \frac{\partial X}{\partial \dot{X}} (\frac{\kappa_1}{r} \sin(\theta) - \frac{\kappa_2}{r^2} \dot{\theta}) &= \dot{X} \\
\frac{\partial \dot{X}}{\partial \theta} + \frac{\partial \dot{X}}{\partial \dot{\theta}} (-2\dot{\theta}) + \frac{\partial \dot{X}}{\partial X} (-2\dot{X}) + \frac{\partial \dot{X}}{\partial \dot{X}} (\frac{\kappa_1}{r} \sin(\theta) - \frac{\kappa_2}{r^2} \dot{\theta}) &= \dot{\theta}^2 - \kappa (X - r_0) - g \cos(\theta).
\end{align*}
\]

(42)

where \(X(\theta, \dot{\theta})\) and \(\dot{X}(\theta, \dot{\theta})\) are the invariant manifold parametrization. We investigate the evaluation of the invariant manifold parametrization through the polynomial Galerkin method, as introduced in Sec. III. Upon their ability of locally approximating functions, polynomials have the advantage of being close w.r.t. derivation, product and sum. Thus, the solution can be found by i) approximating the dynamics as a reduced order polynomial, and guessing a polynomial form for \(X\) and \(\dot{X}\), ii) substituting in (42), and evaluating the free parameters in \(X\) and \(\dot{X}\) such that same order terms are equal.

\[
\begin{align*}
\tau_i(x_1, \dot{x}_1, x_2, \dot{x}_2, \ldots, x_n, \dot{x}_n) &\equiv 0. \text{ So they fulfill our main goal, i.e. that the system at steady state evolves in open loop without any injection of external energy.}
\end{align*}
\]
Given a smooth function \( f : \mathbb{R}^{2n} \to \mathbb{R}^n \), it is well known that its polynomial approximation can be evaluated as multivariate Taylor expansion. We consider here terms up to the third order

\[
\begin{align*}
\dot{\theta} &\simeq -2r\dot{\theta}\left(\frac{1}{r_0} - \frac{1}{r_0^2} (r - r_0)\right) + g\left(\frac{1}{r_0} - \frac{1}{r_0^2} (r - r_0) + \frac{1}{r_0^3} (r - r_0)^2 \theta - \frac{\kappa}{r_0^2} \theta^3\right) \\
&\quad - \kappa_1 \left(\frac{1}{r_0} - \frac{1}{r_0^2} (r - r_0) + \frac{1}{r_0^3} (r - r_0)^2 \theta\right) \\
\dot{r} &\simeq +r\dot{\theta}^2 - g\left(1 - \frac{g^2}{r_0^2}\right) - \kappa_2 (r - r_0) .
\end{align*}
\]

(43)

Regarding instead the maps \( X \) and \( \dot{X} \), we start by considering the symmetry of Eq. (41), in \( \theta \) w.r.t. 0. Indeed it is easy to realize that if \((-\dot{\theta}, -\dot{\theta}, \dot{\theta}, \dot{\theta})\) is a system evolution, than also \((-\dot{\theta}, -\dot{\theta}, \dot{\theta}, \dot{\theta})\) verifies the dynamics. This implies that \( X \) and \( \dot{X} \) must be even, i.e. that their polynomial approximation does not present odd terms. As trade-off between complexity and accuracy we consider a forth order polynomial

\[
\begin{align*}
X(\theta, \dot{\theta}) &\equiv \begin{bmatrix} r_0 - \frac{g}{r_0^2} \\ 0 \end{bmatrix} + S(\theta, \dot{\theta}) \begin{bmatrix} \theta \\ \dot{\theta} \end{bmatrix} + F(\theta, \dot{\theta}) \begin{bmatrix} \theta \\ \dot{\theta} \end{bmatrix} ,
\end{align*}
\]

(44)

with

\[
S(\theta, \dot{\theta}) = \begin{bmatrix} a_1 \theta + a_2 \dot{\theta} \\ a_3 \theta + a_4 \dot{\theta} \end{bmatrix}
\]

\[
F(\theta, \dot{\theta}) = \begin{bmatrix} a_{10} \theta^3 + a_{13} \theta \dot{\theta}^2 + a_{12} \theta^2 \dot{\theta} + a_{11} \dot{\theta}^2 + a_{14} \dot{\theta}^3 \\ b_{10} \theta^3 + b_{13} \theta \dot{\theta}^2 + b_{12} \theta^2 \dot{\theta} + b_{11} \dot{\theta}^2 + b_{14} \dot{\theta}^3 \end{bmatrix},
\]

where \(a_i, b_i\) with \(i \in \{3,4,5,10,11,12,13,14\}\) are the free parameters defining the geometry of the manifold. Introducing (43) and (44) into (42) yields 16 algebraic equations in the unknowns \(a_i, b_i\), that we report in Appendix C. To solve them standard symbolic solvers can be employed (we used \textit{solve} from \textit{MatLab}). The solutions exist in closed form and they are ratios of two multivariate polynomial of the 15th order in the parameters \(\kappa_1, \kappa_2, g, r_0\). We can not report their general form here for the sake of space, however Appendix D presents the case in which \(r_0 = 1m\) and \(g = 9.81 \text{ m/s}^2\). We obtain the mode dynamics by substituting the two maps in the \(\dot{\theta}\) dynamics

\[
\dot{\theta} = -2 \frac{\ddot{X}(\theta, \dot{\theta})}{X(\theta, \dot{\theta})} \theta + \frac{g X(\theta, \dot{\theta}) \sin(\theta) - \kappa_1 \theta}{X(\theta, \dot{\theta})^2} .
\]

(45)

where the dependency of \( X \) and \( \dot{X} \) on \( \kappa_1, \kappa_2, r_0, g \) is omitted.

\section*{C. Simulative Results}

In order to obtain a convex oscillation (i.e. running-like) for all the considered angular velocities \(\dot{\theta}\), we set \(\kappa_1 = 20 \frac{1}{s^2}\) and \(\kappa_2 = 60 \frac{1}{s^2}\). Fig. 8 presents the modal trajectories of the system, superimposed at the corresponding tangent force field (i.e. (45)). Fig 9(a) and Fig. 9(b) present the manifold (i.e. respectively the maps \( X \) and \( \dot{X} \)), on which the orbits corresponding to the initial condition \(\theta = 0\) and \(\dot{\theta} = \frac{\pi}{2} \text{ rad/s}\) are drawn in solid line.

We consider now the application of the control law (17) to stabilize the manifold parametrized by \( X \) and \( \dot{X} \). Note that, since the system has two degrees of freedom \( \Sigma \) and \( \Gamma \) (14) are scalars

\[
\Sigma = \kappa_2 - \dot{\theta}^2 , \quad \Gamma = 0 .
\]

(46)

Thus the second condition of Theorem 1 can be applied. We consider here a pure damping feedback on the slave variable \( r \), resulting in the following control law

\[
\begin{align*}
\tau_1 &= -2 \frac{X(\theta, \dot{\theta}) - X(\theta, \dot{\theta})^r}{X(\theta, \dot{\theta})^r} \dot{\theta} + \frac{g r - X(\theta, \dot{\theta}) r}{X(\theta, \dot{\theta})^r} \sin(\theta) \\
&\quad - \kappa_1 \frac{r - X(\theta, \dot{\theta}) r}{X(\theta, \dot{\theta})^r} \\
\tau_2 &= \kappa_2 (r - \dot{X}(\theta, \dot{\theta})) ,
\end{align*}
\]

(47)

which, from (19), stabilizes the manifold if

\[
\kappa_2 > \dot{\theta}^2 , \quad \kappa_2 > \frac{2\dot{\theta}^2}{\dot{\theta}^2 - \kappa_2} ,
\]

(48)
Figure 10. Inverted elastic pendulum (41) controlled through (17), with $\kappa_d = 1$ (a,b,c), $\kappa_d = 10$ (d,e,f). The considered physical parameters are $\kappa_1 = 20\frac{\pi}{s}$, $\kappa_2 = 60\frac{\pi}{s}$, $g = 9.81\frac{m}{s^2}$, $r_0 = 1m$. The selected initial condition is $\theta = 0$, $\dot{\theta} = \frac{\pi}{16}\frac{rad}{s}$, $r \simeq 0.54m$, $\dot{r} = 0.5\frac{N}{m}$. Note that for these values the system is outside the invariant manifold, indeed $r(0) - X(\theta(0), \dot{\theta}(0)) \simeq -\frac{1}{16}m$ and $\dot{r}(0) - X(\theta(0), \dot{\theta}(0)) = -0.5\frac{N}{m}$. Panels (a,d) present the time evolution of the Lagrangian variables $\theta$ and $r$. Panels (b,e) show the control action generated by the controller. Panels (c,f) present the evolution of the mass in Cartesian space.

Figure 11. Evolution of inverted elastic pendulum (41) controlled through (34), with $\kappa_d = 10\frac{Ns}{m}$, $E^- = 21J$, $E^+ = 22J$, $x^- = -\frac{\pi}{16}$, $x^+ = +\frac{\pi}{16}$, $\gamma = 1N$. The system starts from the equilibrium $r = r_0 = \frac{\pi}{16}$, $\theta = 0$. The controller successfully increases the system oscillations, while maintaining the two degrees of freedom synchronized (i.e. on the modal manifold). This is particularly evident in panel (b).

Figure 12. The control actions generated by (34) while controlling (41) are presented in solid line. The consider parameters are $\kappa_d = 10\frac{Ns}{m}$, $E^- = 21J$, $E^+ = 22J$, $x^- = -\frac{\pi}{16}$, $x^+ = +\frac{\pi}{16}$, $\gamma = 1N$, and initial condition $r = r_0 = \frac{\pi}{16}$, $\theta = 0$. The dashed lines indicate the corresponding actions that would have been needed to implement the same behavior in a rigid system.

where $\dot{\theta}$ is defined by (45). Note that all the terms in (47) are polynomial, except for $\sin(\theta)$. This makes the control evaluation relatively fast, and easily implementable in real time.

Fig. 10 presents the evolution of the system (41) controlled by (47), for two different choices of $\kappa_d$: a low gain $\kappa_d = 1\frac{Ns}{m}$, and an higher one $\kappa_d = 10\frac{Ns}{m}$. The initial condition is $\theta = 0$, $\dot{\theta} = \frac{7}{16}\pi$, $r = X(0, \frac{7}{16}\pi) - \frac{3}{16}m \simeq 0.54m$, $\dot{r} = \dot{X}(0, \frac{7}{16}\pi) - 0.5\frac{Ns}{m} = 0.5\frac{Ns}{m}$. In both cases the robot converges to a stable running-like oscillation. Panels (a,d) show the time evolutions. In the less damped case the $r$ variable converges
Figure 13. Cartesian evolution of inverted elastic pendulum (41) controlled through (34), with $\kappa_2 = 10$, $E^- = 21J$, $E^+ = 22J$, $x^- = -\frac{\pi}{2}$, $x^+ = +\frac{\pi}{2}$, $\gamma = 1N$. The system starts from the equilibrium $r = r_0 = \frac{\pi}{2}$, $\theta = 0$. Note that due to the particular choice of $\kappa_1$ and $\kappa_2$ trajectories are convex for all the simulated evolutions.

more slowly and with an overshoot. Looking to the control actions (b,e), this translates into a much more complex action of the decoupling controller $\tau_d$. Note that in both cases the control action converges to zero, i.e. the robot evolves on the manifold following autonomous trajectories. Panels (c,f) show the trajectory of the center of mass in the Cartesian space. In (c) a much more dynamic transient can be observed.

Fig. 11 presents the time evolutions of system (41) controlled through (34). Note that the system is conservative, and the energy

$$E(\theta, \dot{\theta}, r, \dot{r}) = \frac{1}{2} (r^2 \dot{\theta}_t^2 + \dot{r}^2) + \frac{1}{2} (\kappa_1 \theta^2 + \kappa_2 (r-r_0)^2) + g (r \cos(\theta))$$

has closed level curves, thus fulfilling the hypotheses of Theorem 2. We considered $E^- = 21J$, $E^+ = 22J$, $x^- = -\frac{\pi}{2}$, $x^+ = +\frac{\pi}{2}$, $\gamma = 1N$. The system starts at the equilibrium, i.e. $\theta = 0$, $r = r_0 = \frac{\pi}{2}$, $\dot{\theta} = 0.84 \pi$, $\dot{r} = 0$. The orbit excitation controller perturbs the system putting it in oscillation. The system reaches the desired level of energy at about 9s. Note that thanks to the stabilizing controller, evolutions remain synchronized during the whole excitation phase despite the perturbations. This is particularly evident from the zero-crossings of the velocities, Fig. 11 panel (b).

The same figure also highlights a key characteristic of the considered mode: the frequency of oscillation of $\theta$ is half of the frequency of oscillation of $r$. This is a purely nonlinear behavior, made possible by the fact that the parametrization of the manifold $X$ decreases in one direction and increases in the other (see Figs. 9(a) and 9(b)).

For the same simulation, Fig. 12 presents a comparison between the control actions exerted by the proposed controller ($\tau_d$ and $\tau_r$), and the ones that would have been necessary to regulate an equivalent rigid robot along the same trajectory ($\tau_{d,0}$ and $\tau_{r,0}$). For all the evolution the soft robot controlled through the proposed controller visibly outperform the rigid counterpart. The forces are in the worst case around one order of magnitude less than the control actions needed to a non elastic version of the robot to implement the same trajectory, as shown by Fig. 12(b). Finally, Fig. 13 presents the Cartesian evolution of the center of mass.

VI. EXPERIMENTAL RESULTS:
OSCILLATORY CONTROL OF A SEGMENTED LEG

As a first experimental validation of the proposed strategy, we consider the soft segmented leg in Fig. 14. We are interested here in generating swing oscillations of its center of mass, in analogy to what was obtained in simulation. The segmented leg is composed by two links of the same length $a$, considered here massless, and a main body, with mass $m$. The leg is mechanically constrained to evolve on the sagittal plane, and the main body to remain vertical. We hypothesize infinite friction between the foot and the environment. Thus the configuration of the robot can be described by the two angles $q_1$ and $q_2$ in Fig. 14.

We consider the following change of coordinates,

$$\theta = \frac{q_1 + q_2}{2}$$

$$r = a \sqrt{2(1 + \cos(q_1 - q_2))},$$

where $\theta$ and $r$ are the polar coordinates of the center of mass w.r.t. the foot position. The resulting dynamics has the SLIP-like form (see [30] for the detailed derivation)

$$\dot{\theta} = -\frac{1}{r} \dot{r} \theta$$

$$\dot{\theta}_t = +r \dot{\theta}^2 - g \cos(\theta) - \kappa \frac{1}{\sqrt{4a^2 - r^2}} (\rho(r) - \rho(r_0)) + \tau_r$$

where $\rho(r) = \arccos(1 - \frac{r^2}{4a^2})$, $\kappa$ is the stiffness of both springs, $\tau_d$ and $\tau_r$ are the control actions, and the other terms are as in (41). The maps $X(\theta, \dot{\theta})$ and $X(\theta, \dot{\theta}_t)$ can be evaluated as in the SLIP case (see Sec. V-B), and they are not reported here for the sake of space.

Note that the physical system presents several non ideal effects making its control challenging, as e.g. actuator dynamics, contact with the ground, non zero weight of the legs, inexact identification of system parameters, neglected friction effects. Furthermore, the angular velocity $\dot{\theta}$ could not be directly measured, and it was instead estimated through a simple high pass filter.

Though it is beyond the scope of this paper to investigate robustness analytically, we empirically test it here with this example, where we do not assume any knowledge of the dynamic form in the decoupling control. We instead consider only the two main ingredients of the proposed control strategy (34), i.e. an excitatory action in the direction of the master variable $\theta$, and a damping action along the slave direction $r$

$$\tau(\theta, \dot{\theta}) = \frac{\tau_1(\theta, \dot{\theta})}{10(\dot{r} - X(\theta, \dot{\theta}))},$$

where $\tau_1$ is as in (35). Empirically we considered a high gain for the damping controller, since in the simulative case this
translated into a small authority of the decoupling controller neglected here.

As in simulation, we consider the target energy level $E^− = 21J, E^+ = 22J$. We performed experiments for five different values of the orbit excitation gain $\alpha$ in $\tau_1$: 0.2Nm, 0.3Nm, 0.5Nm, 0.7Nm, 0.9Nm. Due to friction, the desired level of energy could not be reached. Instead energy injected through $\tau_1$ is compensated by the dissipation and a different equilibrium is reached for each value of $\alpha$. Fig. 15 presents the photo-sequences of a single oscillation for two of the considered gains. Fig. 16 presents the evolution of $\theta$ and $r$ for $\alpha = 0.5$Nm. Control action is turned on at $0s$, and it takes about $2s$ to bring the system on a stable oscillation. It is worth noticing that the resulting nonlinear oscillation ($\theta, r$) is actually very close to the ideal one ($\theta, X(\theta, \dot{\theta})$), as evidenced by Panel (b) of the same figure.

Fig. 17 shows the center of mass’ evolutions in Cartesian coordinates, for all the considered gains and for a period of $15s$. Note that the bigger is the gain, the larger are the oscillations, i.e. the higher is the energy level reached. The resulting oscillations are slightly concave, and highly repeatable.

These results suggest that the proposed strategy can be used to excite the normal modes of soft robots, generating stable and repeatable nonlinear oscillations also in the presence of many unideal behaviors in the controlled system.

Fig. 18 illustrates the resulting evolutions superimposed to the ideal modal manifold, i.e. to the surface $r = X(\theta, \dot{\theta})$. The matching is good, with larger discrepancies for high speeds and positive values of $\theta$. The asymmetry of behavior w.r.t. to $\theta$ was already evident in Fig. 16, and it is probably due to the effect of leg dynamics, which is neglected in (51). The error observed for large velocities is instead probably due to the persistent excitation (as theoretically discussed in Sec. IV-C), and to the imperfect knowledge of $\dot{\theta}$. Future work will be devoted to more in-depth analysis of the theoretical implications of these aspects.

Finally, the algorithm was preliminarily tested on a quadruped built using four of the above discussed soft segmented legs. The proposed control strategy is able to excite stable natural oscillations also in this more complex system, as shown in Fig. 19. Future work will be devoted to the exploitation of such oscillations in performing efficient locomotion patterns.

VII. CONCLUSIONS

Soft Robots are robotic systems in which elastic elements are purposefully introduced in the mechanical structure. It is thus intuitively clear that soft robots should be able to perform oscillatory tasks with good efficiency. We formalized this intuition in the linear case in Sec. II. However, the nonlinearity of the robot dynamics make the problem of studying and generating oscillations non-trivial. Classical techniques can regulate the system on a specific trajectory or limit cycle by a partial or complete cancellation of the dynamics, thus defeating the purpose of introducing springs. In this work we took inspiration from the natural world and from nonlinear dynamical system theory, proposing to generate very efficient oscillations in soft robots by regulating the system on a nonlinear extension of a linear eigenspace. This way the natural dynamics of the system is fully exploited. After a brief survey about the nonlinear normal mode theory, we moved to the mode stabilization in the nonlinear case. We also considered the problem of exciting a specific set of trajectories on the manifold. We then discussed the analysis problem, proposing an approximated analytic solution for the SLIP model. Simulations were presented to show the effectiveness of the method. Finally, leveraging on insights gained from simulations, an empirical simplification of the proposed controller is used to experimentally induce nonlinear oscillations in a segmented leg. In addition to showing the practical feasibility and robustness of the method, experiments served to understand its main practical limitations, which we briefly discussed at the end of the same section.

Many are the aspects of this work that will require further investigation in the future. For what concerns the control part, future work will be devoted to better understanding the role of the decoupling controller, and to the possibility of generating persistent oscillatory actions without sensibly changing the manifold shape. e.g. by exciting nonlinear resonances (as it happens in the linear case). Regarding the analysis problem, application to multi-DoF soft robotics systems calls for the development of novel tools and techniques for NNM analytical derivation. Finally, from an experimental point of view our work will focus on implementing stable natural oscillations in more complex systems and meaningful tasks, as the execution of locomotion gates with legged robots, oscillatory pick and place with multi-DoF soft arms, etc.

APPENDIX A

We present in this section the proof of Proposition 1.
Figure 15. Photo-sequences of nonlinear oscillations induced by the proposed algorithm on a segmented soft leg. Panels (a-e) present one oscillation for \( \alpha = 0.3 \text{Nm} \), while panels (f-j) show the case of \( \alpha = 0.9 \text{Nm} \).

Figure 16. Experimental evolutions of master variable \( \theta \) and slave variable \( r \), for \( \alpha = 0.5 \text{Nm} \). After an initial transient in which the algorithm injects energy, the segmented leg starts to evolve according to a stable nonlinear oscillation. Panel (b) also presents the ideal evolution on the manifold \( X(\theta, \dot{\theta}) \) of the slave variable \( r \) for the measured evolution of \( \theta \).

Figure 17. Experimental trajectories in Cartesian coordinates of the segmented leg’s center of mass controlled through (52), with different values of the energy injection gain \( \alpha \). Stable oscillations result for all the considered gains.

Figure 18. Experimental evolutions represented in the space \((\theta, \dot{\theta}, r)\), obtained for several choices of the gain \( \alpha \) and 15s of oscillations. We also superimpose the ideal modal manifold \( r = X(\theta, \dot{\theta}) \). The matching is good for all the considered gains, despite the many unideal characteristics of the system.

Proof. Let's express (1) in state form by considering the state vector \( \begin{bmatrix} p^T & v^T \end{bmatrix}^T = \begin{bmatrix} x^T & \dot{x}^T \end{bmatrix}^T \)  
\[
\begin{bmatrix} \dot{p} \\ \dot{v} \end{bmatrix} = \begin{bmatrix} 0 & I \\ -M^{-1}K & 0 \end{bmatrix} \begin{bmatrix} p \\ v \end{bmatrix} + \begin{bmatrix} 0 \\ M^{-1}F \end{bmatrix}. \tag{53}
\]

Consider the orthonormal matrix \( V \in \mathbb{R}^{n \times n} \), such that \( J = V^T K V \) is in Jordan form. By considering the following change
of coordinates,
\[
\begin{bmatrix}
\pi \\
v
\end{bmatrix} =
\begin{bmatrix}
V & 0 \\
0 & V
\end{bmatrix}
\begin{bmatrix}
p \\
v
\end{bmatrix}
\tag{54}
\]
the dynamics (53) becomes
\[
\begin{bmatrix}
\pi \\
v
\end{bmatrix} =
\begin{bmatrix}
0 & 1 \\
-J & 0
\end{bmatrix}
\begin{bmatrix}
\pi \\
v
\end{bmatrix} +
\begin{bmatrix}
0 \\
f
\end{bmatrix}
\tag{55}
\]
where \( f = V^T M^{-1} \tau \). This is equivalent to a set of decoupled systems in the form
\[
\begin{bmatrix}
\pi_i \\
v_i
\end{bmatrix} =
\begin{bmatrix}
0 & I_{m_i \times m_i} \\
-J_i & 0
\end{bmatrix}
\begin{bmatrix}
\pi_i \\
v_i
\end{bmatrix} +
\begin{bmatrix}
0 \\
f_i
\end{bmatrix},
\tag{56}
\]
where \( \pi_i, v_i, f_i \in \mathbb{R}^{m_i} \) are elements of \( \pi, v, f \) related to the \( i \)-th eigenspace. \( J_i \) is the \( i \)-th diagonal block of \( J \).

Consider now the following relationship
\[
K^{1/2} (M^{-1} K)^{1/2} = K^{1/2} M^{-1} K^{1/2} = (K^{1/2} M^{-1} K^{1/2})^T
\tag{57}
\]
where we exploited the hypothesis \( K > 0 \) to define \( K^{1/2} \), and the symmetry of both \( K \) and \( M \) in the second step. Eq. (57) tells us that \( M^{-1} K \) is similar to a symmetric matrix. Thus all its eigenvalues are real.

Note also that \( \forall v \in \mathbb{R}^{m_i} \)
\[
v^T K^{1/2} M^{-1} K^{1/2} v = (K^{1/2} v)^T M^{-1} (K^{1/2} v) > 0,
\tag{58}
\]
where in the last step we exploited that \( M > 0 \Rightarrow M^{-1} > 0 \). Thus, \( M^{-1} K \) is similar to a positive and symmetric matrix, and therefore all its eigenvalues are also positive.

This implies that \( J_i \) has the form
\[
J_i =
\begin{bmatrix}
\lambda_i & 1 & 0 & \cdots & 0 \\
0 & \lambda_i & 1 & \cdots & 0 \\
\vdots & \vdots & \vdots & \ddots & \vdots \\
0 & 0 & 0 & \cdots & \lambda_i \\
0 & 0 & 0 & \cdots & \lambda_i
\end{bmatrix},
\tag{59}
\]
with \( \lambda_i > 0 \). This is in turn implies that through simple permutations the system (56) can be expressed as a series of \( m_i \) linear oscillators
\[
\begin{bmatrix}
\pi_{i,j} \\
v_{i,j}
\end{bmatrix} =
\begin{bmatrix}
0 & 1 \\
-\lambda_i & 0
\end{bmatrix}
\begin{bmatrix}
\pi_{i,j} \\
v_{i,j}
\end{bmatrix} +
\begin{bmatrix}
0 \\
f_{i,j}
\end{bmatrix} -
\begin{bmatrix}
0 \\
\pi_{i,j-1}
\end{bmatrix},
\tag{60}
\]
where \( \pi_{i,j}, v_{i,j}, f_{i,j} \in \mathbb{R} \) are the \( j \)-th elements of \( \pi_i, v_i, f_i \in \mathbb{R}^{m_i} \). Note that series of linear systems are stable if each subsystem is stable. Thus to stabilize (56) it is sufficient to damp each oscillator separately. At this end we can use
\[
f_i = -\beta v_i,
\tag{61}
\]
where \( \beta \in \mathbb{R}^+ \) is a strictly positive constant.

So, to make attractive the eigenspace associated to the first eigenvector it will be sufficient to regulate at 0 the dynamics expressed on all the other eigenspaces. This can be done by applying (61) for all \( i \neq 1 \)
\[
f = -\beta
\begin{bmatrix}
0_{m_1 \times m_1} & 0_{m_1 \times (n-m_1)} \\
0_{(n-m_1) \times m_1} & I_{(n-m_1) \times (n-m_1)}
\end{bmatrix}
v.
\tag{62}
\]
In the original coordinates (62) is
\[
\tau = -\beta MV
\begin{bmatrix}
0_{m_1 \times m_1} & 0_{m_1 \times (n-m_1)} \\
0_{(n-m_1) \times m_1} & I_{(n-m_1) \times (n-m_1)}
\end{bmatrix} V^T \xi.
\tag{63}
\]

\section*{Appendix B}

In this appendix we propose two lemmas about the stability of linear time variant mechanical system, which are instrumental to the proof of Theorem 1. Let's consider the system
\[
\begin{bmatrix}
\dot{p} \\
\dot{v}
\end{bmatrix} =
\begin{bmatrix}
0 & I \\
-K(t) & -D(t)
\end{bmatrix}
\begin{bmatrix}
p \\
v
\end{bmatrix},
\tag{64}
\]
where \( p \in \mathbb{R}^n \) is the system configuration space, \( v \in \mathbb{R}^n \) are the velocities. \( K(t) \in \mathbb{R}^{n \times n} \) and \( D(t) \in \mathbb{R}^{n \times n} \) respectively the time-varying stiffness and damping.

\textbf{Lemma 1.} A \( \delta < \infty \) always exists such that if \( \|K(t)\| < \delta, \|D(t)\| < \delta, \rho(K(t)) > 0, \) and \( \rho(D(t)) > 0, \) then (64) is stable.

\textbf{Proof.} We prove here the thesis in the case of Frobenious norm. This is without loss of generality since any other standard matrix norm can be upper and lower bounded by the Frobenious norm [56].

Classic results [57] assure that always exists a \( \gamma \in \mathbb{R}^+ \) such that if
\[
\left\| \frac{d}{dt} \begin{bmatrix}
0 & I \\
-K(t) & -D(t)
\end{bmatrix} \right\|_F < \gamma
\tag{65}
\]
and
\[
\rho \left( \begin{bmatrix}
0 & I \\
-K(t) & -D(t)
\end{bmatrix} \right) < 0 \quad \forall t,
\tag{66}
\]
then (64) is stable.

The first condition is assured by considering \( \gamma = \sqrt{2}\delta \)
\[
\left\| \frac{d}{dt} \begin{bmatrix}
0 & I \\
-K(t) & -D(t)
\end{bmatrix} \right\|_F^2 = \left\| \begin{bmatrix}
0 & 0 \\
-K(t) & -D(t)
\end{bmatrix} \right\|_F^2
\tag{67}
\]
\[
\leq 2 \delta^2 = \gamma^2,
\]
which leads to (66) by extracting the square root.

It is worth noticing that condition (66) is equivalent to considering the stability of all the possible time invariant
systems obtained by fixing the time in the dynamic matrix. Thus we can prove condition (66) by considering that a sufficient condition for the stability of a linear mechanical system is that its stiffness and damping matrices are both positive definite. This is assured by the hypotheses $\rho(K(t)) > 0$, and $\rho(D(t)) > 0$.

**Lemma 2.** If $M(t) \in \mathbb{R}^{n \times n}$ exists such that $M^T K M$ and $M^T D M$ are both diagonal \forall t, and if

$$\lambda_i(K(t)) > 0, \quad \lambda_i(D(t)) > -\frac{\lambda_i(K(t))}{\lambda_i(D(t))} \quad \forall t \quad (68)$$

where $\lambda_i(\cdot)$ extracts the eigenvalue corresponding to the $i$-th column of $M(t)$, then (64) is stable.

**Proof.** By considering the following change of coordinates,

$$\begin{pmatrix} \pi \\ u \end{pmatrix} = \begin{bmatrix} M(t) & 0 \\ 0 & M(t) \end{bmatrix} \begin{pmatrix} \rho \\ v \end{pmatrix}$$

the dynamics becomes equivalent to a set of decoupled undimensional time variant oscillators

$$\begin{pmatrix} \dot{\pi}_i \\ \dot{u}_i \end{pmatrix} = \begin{bmatrix} 0 & 1 \\ -\lambda_i(K(t)) & -\lambda_i(D(t)) \end{bmatrix} \begin{pmatrix} \pi_i \\ u_i \end{pmatrix} .$$

The stability of such system under the bounds (68) can be proven using the Lyapunov candidate

$$V(\pi_i, u_i, t) = \pi_i^2 + \frac{v_i^2}{\lambda_i(K(t))},$$

and the Barbalat lemma. For the sake of conciseness we do not report here the whole derivation, that can be found in [58].

**APPENDIX D**

We present here the coefficients of the manifold parametrization $X(\hat{\theta}, \hat{\theta})$ and $\dot{X}(\hat{\theta}, \hat{\theta})$ derived in Sec. V in the form described by (44), when $r_0 = 1m$ and $g = 9.81 \frac{m}{s^2}$

$$a_i(\kappa_1, \kappa_2) = -\frac{\alpha_i(\kappa_1, \kappa_2)}{\gamma_i(\kappa_1, \kappa_2)} \quad (71)$$

$$\alpha_3 = 5 \left(-400 d_1 \kappa_2 + 3.910^3 d_1 \kappa_2 + 2.10^3 d_1 \kappa_2 + 999 \kappa_2^2 \right) \quad (72)$$

$$\gamma_3 = 10^3 \kappa_2^2 \left(4d_1 + \kappa_2 \right) \quad (73)$$

$$\alpha_5 = -20 d_1 + 20 \kappa_2 - 2d_1 \kappa_2 - \kappa_2^2 \quad (74)$$

$$\gamma_5 = \kappa_2^2 \left(4d_1 + \kappa_2 \right) \quad (75)$$

**APPENDIX C**

We report in the following the 16 algebraic equations resulting from the evaluation of Eq. (9) for the system (41)

$$\begin{align*}
    b_{10} - a_{13} d_1 + a_4 d_6 &= 0 \\
    b_{13} - 4a_{10} - 2a_{12} d_1 + a_4 d_5 + 2a_5 d_6 &= 0 \\
    b_{12} - 3a_{13} - 3a_{11} d_1 + 2a_5 d_5 + 4a_4 d_7 &= 0 \\
    b_3 - a_4 d_1 &= 0 \\
    b_{11} - 2a_{12} - 4a_{14} d_1 + 2a_5 d_7 + 2a_4 b_5 d_3 &= 0 \\
    b_4 - 2a_3 - 2a_5 d_1 &= 0 \\
    b_{14} - a_1 + 4a_5 b_3 d_3 &= 0 \\
    b_5 - a_4 &= 0 \\
    b_4 d_6 - b_{13} d_1 - a_{10} \kappa_2 &= 0 \\
    b_4 d_5 - a_{13} \kappa_2 - 2b_{12} d_1 - 4b_{10} + 2b_5 d_6 &= 0 \\
    a_{3} - 3b_{13} - a_{12} \kappa_2 - 3b_{11} d_1 + 2b_5 d_5 + b_4 d_7 &= 0 \\
    g - 2a_3 \kappa_2 - 2b_4 d_1 &= 0 \\
    a_{4} - 2b_{12} - a_{11} \kappa_2 - 4b_{14} d_1 + 2b_5 d_7 + 2a_4 b_5 d_3 &= 0 \\
    2b_3 + a_4 \kappa_2 + 2b_5 d_1 &= 0 \\
    a_5 - b_{11} - a_{14} \kappa_2 + 4b_5^2 d_3 &= 0 \\
    r_0 - b_4 - a_5 \kappa_2 - a_5 \kappa_2 &= 0
\end{align*} \quad (69)$$

where

$$\begin{align*}
    d_1 &= g \left( \frac{1}{r_0} + \frac{g}{\kappa_2 r_0} + \frac{g^2}{\kappa_2^2 r_0^2} \right) - k_1 \left( \frac{1}{r_0} + \frac{g}{\kappa_2 r_0} + \frac{g^2}{\kappa_2^2 r_0^2} \right) \\
    d_2 &= \frac{1}{r_0} + \frac{2g}{\kappa_2 r_0} \\
    d_3 &= \frac{1}{r_0} + \frac{g}{\kappa_2 r_0} \\
    d_4 &= \frac{2}{r_0} + \frac{6g}{\kappa_2 r_0} \\
    d_5 &= 2b_3 d_3 + g a_4 d_2 - k_1 a_4 d_4 \\
    d_6 &= g a_3 d_3 - k_1 a_3 d_4 + \frac{g}{r_0} \\
    d_7 &= 2b_4 d_3 + g a_5 d_2 - k_1 a_5 d_4
\end{align*} \quad (70)$$

$$\begin{align*}
    \alpha_{12} &= 1.210^5 d_1 \kappa_3^3 - 2.710^4 d_1 \kappa_3 + 144 d_1 \kappa_3^4 \\
    &- 2.6010^4 d_1 \kappa_3^3 + 2.510^3 d_1 \kappa_2^3 + 788 d_2 \kappa_2^4 + 399 d_3 \kappa_2^4 \\
    &- 788 \kappa_3^4 + 29 \kappa_3^5 + 1.510^3 d_1^2 \kappa_2^3 + 688 d_2 \kappa_2^3 + 310^3 d_1 \kappa_2^2 \\
    &- 28 d_1^2 \kappa_2^4 - 199 d_1^3 \kappa_3^3 - 277 d_1^4 \kappa_3^2 - 9.210^3 d_1^2 \kappa_2^2 \\
    &+ 488 d_1^3 \kappa_2^3 + 1.310^3 d_1^3 \kappa_2^2 - 633 d_1^2 \kappa_2^3 \\
    &- 3.110^3 d_1^2 \kappa_3^3 + 64 d_1^3 \kappa_3^2 + 177 d_1^4 \kappa_3^2 \\
    &+ 4.510^4 d_1^2 \kappa_3^2 - 4.610^3 d_1 d_2 \kappa_3^2 - 2.510^4 d_1^3 d_2 \kappa_2 \\
    &+ 177 d_1 d_2 \kappa_3^4 - 4.610^3 d_1 d_3 \kappa_3^3 + 1.510^4 d_1^2 d_3 \kappa_2 \\
    &- 1.210^4 d_1 d_3 \kappa_4 + 400 d_1 d_3 \kappa_4 + 2.710^3 d_1^3 \kappa_1 \kappa_3 \\
    &- 78 d_1^3 \kappa_4^2 + 4.610^3 d_1^2 d_2 \kappa_1 \kappa_3 + 488 d_1 d_3 \kappa_1 \kappa_3 \\
    &+ 2.510^4 d_1^3 \kappa_1 \kappa_3 - 161 d_1 d_3 \kappa_1 \kappa_3^4 + 944 d_1^2 d_4 \kappa_1 \kappa_3^2 \\
    &- 48.10^3 d_4 \kappa_1 \kappa_3^3 - 133 d_1^3 d_4 \kappa_1 \kappa_3^2
\end{align*} \quad (76)$$

$$\begin{align*}
    \gamma_{12} &= 2 \kappa_2^4 \left(4 d_1 + \kappa_2 \right)^2 \left(16 d_1 + \kappa_2 \right) \quad (77)
\end{align*}
\[ \alpha_{14}(\kappa_1, \kappa_2) = 2 \cdot 10^5 d_1 d_2^2 + 9.2 \cdot 10^3 d_1^3 d_3 - 1.5 \cdot 10^3 d_1^2 \kappa_2 - 566 d_1 \kappa_3^3 + 3.4 \cdot 10^4 d_2 \kappa_2^2 - 4.2 \cdot 10^3 d_1^3 \kappa_2 + 22 d_1 \kappa_4 - 1.5 \cdot 10^3 d_1 \kappa_3 + 277 d_1^2 \kappa_2 + 20 d_2 \kappa_4 - 1.2 \cdot 10^3 d_3 \kappa_3^2 + 39 d_3 \kappa_4^2 - 2.5 \cdot 10^3 d_1^4 + 788 \kappa_3^5 - 49 \kappa_2^4 + \kappa_5^2 - 2.3 \cdot 10^2 \kappa_2^2 + 133 d_1^2 \kappa_2 + 300 d_1 \kappa_3^2 + 877 d_1^2 d_2 \kappa_2 - 2.2 \cdot 10^3 d_1^3 d_2\kappa_3 + 80 d_1^2 d_3 \kappa_3^2 + 96 d_1^3 d_3 \kappa_3^2 + 1.4 \cdot 10^4 d_1 d_2 \kappa_3 - 1.1 \cdot 10^4 d_1^2 \kappa_2^2 - 2.91 \cdot 10^3 d_1^3 d_2 \kappa_2 + 244 d_1^2 d_2 \kappa_3^2 - 6.9 \cdot 10^3 d_1 d_3 \kappa_3^2 + 1.4 \cdot 10^4 d_1^2 d_3 \kappa_3 - 2.1 \cdot 10^4 d_1^2 d_4 \kappa_1 + 78 d_1 d_3 \kappa_4^2 - 1.9 \cdot 10^4 d_1 d_3 \kappa_4 + 8 d_1 d_3 \kappa_2^4 - 3.5 \cdot 10^3 d_1 d_4 \kappa_1 \kappa_2 + 177 d_1 d_3 \kappa_3^3 - 2 d_1 d_3 \kappa_4^4 + 110 d_1^3 d_1 \kappa_1 \kappa_3 + 310 d_1^2 d_2 \kappa_2 - 24 d_1 d_3 \kappa_3^3 - 88 d_1^2 d_4 \kappa_1 \kappa_2^2 - 1.4 \cdot 10^3 d_1 d_4 \kappa_1 \kappa_2^2 \] (78)

\[ \gamma_1(\kappa_1, \kappa_2) = \kappa_3^3 (4 d_1 + \kappa_2^2) (64 d_1^2 + 20 d_1 d_2 + \kappa_2^2) \] (79)

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