Coupling of integral methods and CFD for modeling complex industrial accidents

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Abstract

Safety enhancement of operations in the chemical and petrochemical industry requires for advances in the tools aimed at supporting risk estimation and evaluation. In conventional risk studies, consequence assessment is carried out through simplified tools and conservative assumptions, often resulting in overestimation of accident severity and worst-case scenarios. Computational Fluid Dynamics (CFD) may overcome the limitation of simplified approaches supporting the study of the dynamic evolution of accidental scenarios and, eventually, the consequences analysis of major accidents. However, the complexity of the problem makes the simulations too computationally demanding; hence an interesting approach is to couple simplified tools based on integral models and CFD. This work is aimed at modeling a safety critical scenario, i.e. domino effect triggered by fire. An integral model is adopted to reproduce a large-scale pool fire, thus simulating the radiative heat received by an exposed pressurized vessel. The behavior of the latter is then modeled through CFD, to investigate the heat-up process and the consequent pressure build up. Potential benefits and limitations of coupling distributed and

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integral models to support consequence assessment studies are discussed. *Keywords:* Volume Of Fluid, liquid stratification, pool fire, safety, major accident hazard, Computational Fluid Dynamics

1 1. Introduction

Safety enhancement of operations in the chemical and petrochemical in-2 dustry requires for advances in the tools aimed at supporting risk estimation and evaluation. Risk analysisuses engineering and mathematical techniques (Crowl and Louvar (2011)), to evaluate consequences of accidents and thus 5 their potential impact (Mannan (2012), Center of Chemical Process Safety 6 (2000)). As remarked by several authors (e.g., Kalantarnia et al. (2009), Landucci and Paltrinieri (2016)), in the consolidated procedures for quantitative risk assessment (QRA), conservative simplifications and a rather static 9 approach are adopted for the consequence assessment of fires/explosions (I 10 et al. (2009)) or toxic dispersion and contaminations (Segu et al. (2014)). 11 This is due to the high number of potential scenarios and uncertainties re-12 lated to accident identification and characterization. However, neglecting the 13 transient and dynamic effects associated with the complex accident evolution 14 may lead to inaccurate estimation of the risk. Updates and implementation 15 of continuously changing quantities, the mitigation effect of safety barriers, 16 and eventually knowledge and evidence on hazard dynamic evolution need 17 to be accounted for a more accurate accident scenario simulation and, thus, 18 for the risk estimation (Villa et al. (2016), Xin et al. (2017), Zarei et al. 19 (2017)). Cascading events represent a critical safety issue characterized by a 20 complex dynamic evolution (Khakzad and Reniers (2015)) and may consti-21

tute high-consequence chains of accidents (Darbra et al. (2010), Reniers and Cozzani (2013)). In case of a cascading effect, a primary accident, such as a fire occurring in a primary unit, propagates to neighboring units triggering secondary accidents in the surrounding plant area, with potential amplification of consequences (Necci et al. (2015)).

Commonly applied approaches for the safety and risk assessment of this type 27 of scenarios are not yet consolidated and are based on strong simplifications. 28 As reported by Alileche et al. (2015), damage and escalation thresholds are 29 commonly applied to identify secondary scenarios, possibly resulting from a 30 domino effect. The results of consequence analysis models, applied to the 31 simulation of primary scenarios, are compared to the threshold values, iden-32 tifying a maximum credible escalation radius (Cozzani et al. (2007)) and 33 performing a screening of escalation events (Cozzani et al. (2013)). This 34 type of screening is important to assess the credibility and the criticality of 35 different escalation scenarios, but the detailed analysis of critical units re-36 quires more advanced tools, such as distributed parameters models. 37

Computational Fluid Dynamics (CFD) modeling is a consolidated tool to 38 support industrial projects development and was recently adopted in the 39 framework of consequence assessment and safety studies (Schmidt (2012), 40 Landucci et al. (2016b)). The advanced features of CFD models make them 41 a promising tool to support the assessment of complex accidental scenar-42 ios, such as three-dimensional pool fires, jet fires and the possible induced 43 cascading events. Such features correspond to: handling complex threedimensional geometries and environments (e.g. Pontiggia et al. (2010), Pon-45 tiggia et al. (2011), Derudi et al. (2014)), analyzing turbulent reactive or

⁴⁷ non-reactive flow of compressible or non-compressible fluids (e.g. Ferziger
⁴⁸ and Peric (2002), Lomax et al. (2002)) and analyzing multi-phase flows.
⁴⁹ Hence CFD may be used to simulate the thermal load on a process vessel
⁵⁰ due to an accidental fire (Masum Jujuly et al. (2015)) and to investigate the
⁵¹ transient behavior of the stored fluid and structure (Bi et al. (2011), Jang
⁵² et al. (2015)) during heat-up.

Several studies were aimed at simulating industrial fires through CFD based 53 tools (Chenthil et al. (2015), Singh et al. (2014)). Pool fire modeling through 54 CFD has been extensively carried out since the 90's, determining the poten-55 tialities of distributed parameters codes in capturing the effects of bunds, 56 wind profiles and confinement in the determination of flame structure and 57 associated effects (Sinai and Owens (1995)). More recently, Sun et al. (Sun 58 et al. (2015), Sun and Guo (2013)) provided a dynamic LNG pool fire simula-59 tion to estimate mitigation through high expansion foam at different burning 60 times. Several authors proposed pool fire simulations to analyze the poten-61 tial occurrence of cascading events (e.g., Bainbridge and Keltner (1988), Ma-62 sum Jujuly et al. (2015), Siddapureddy et al. (2016)). However, they focused 63 on the determination of the thermal loads distribution on the outer surface 64 of the vessels engulfed by the flames (Siddapureddy et al. (2016)) or exposed 65 to distant source radiation (Masum Jujuly et al. (2015)), while the complex 66 behavior of the tank lading was not taken into account. 67

⁶⁸ Due to the high turbulence, jet fire modeling is also a challenging task that ⁶⁹ was addressed in recent years (Ferreira and Vianna (2016), Hooker et al. ⁷⁰ (2016), Sun et al. (2017), Zhao and Magenes (2016)). Wang et al. (2014) ⁷¹ adopted FireFOAM to study the radiation characteristics of hydrogen and ⁷² hydrogen/methane jet fires, capturing the fluctuations in flame length and ⁷³ radiant fraction. Jang et al. (Jang et al. (2015)) simulated a hydrogen jet ⁷⁴ fire from an accidental leak, determining the dynamic evolution of the flame ⁷⁵ temperature and shape into a complex three-dimensional layout. A real scale ⁷⁶ pipe rack was reproduced, determining the flame impact zone as well as the ⁷⁷ heat radiation profiles. The utilization of CFD to support three-dimensional ⁷⁸ QRA studies is also documented in other studies (e.g., I et al. (2009)).

The analysis of the transient behavior of tanks exposed to either pool or 79 jet fires was developed since the early 70's by the US Federal Railroad Ad-80 ministration and Transport Canada (Johnson (1998b), Johnson (1998a)). 81 Since then, several studies were undertaken, focusing on the thermal re-82 sponse of LPG tanks exposed to fire (Moodie (1988)). Lumped-parameter 83 models (Aydemir et al. (1988), Beynon et al. (1988), Birk (1989), Dancer and 84 Sallet (1990), Graves (1973), Heymes et al. (2013), Johnson (1998b), John-85 son (1998a), Ramskill (1988), Salzano et al. (2003)) represent the simplest 86 modeling approach to the problem, needing limited computational time and 87 set-up parameters but usually neglecting important complicating phenom-88 ena such as the liquid thermal stratification and expansion (Landucci et al. 89 (2016a)). 90

Distributed parameters models were applied to the assessment of similar
problems, e.g. to the analysis of the heat-up of water in pressurized tanks
(Gandhi et al. (2013), Han et al. (2009)), of asphalt in cylindrical tanks
(Costa et al. (2013)) or cryogenic liquids (Das et al. (2004), Ren et al. (2013),
Roh et al. (2013), Wang et al. (2013)) exposed to external heat sources.
Some studies were devoted to the analysis of small scale tanks containing

pressurized hydrogen gas exposed to localized fires, supported by specific 97 experiments (e.g., Zheng et al. (2012), Zheng et al. (2013)). Therefore the 98 experience with CFD tools is limited to the simulation of the dynamic evo-99 lution of fluids with physical and chemical features completely different with 100 respect to LPG and, more in general, to pressurized liquefied hydrocarbons. 101 Only recently CFD models were developed to study the effect of fire exposure 102 on LPG tanks. Bi et al. (2011) considered small-scale LPG tanks, whereas 103 Landucci and coworkers (D'Aulisa et al. (2014), Landucci et al. (2016a)) an-104 alyzed large-scale LPG vessels. However, the simulation set-up did not allow 105 to model complex fire scenario exposure. In fact, the heat load was derived 106 empirically or from literature, considering only symmetric and homogeneous 107 heat flux conditions. Moreover, the adopted computational discretization 108 only allowed to separately tracing the liquid and vapor phases, imposing the 109 initial filling level and simulating in details the sole evolution of the liquid 110 phase. 111

Another key issue that may be investigated through distributed parameters 112 code is the structural response of equipment when exposed to fire. In this 113 case, finite elements modeling (FEM) may be applied for the assessment 114 of the mechanical behavior, thus supporting the prediction of failure condi-115 tions, as documented in several industrial studies (e.g., Andreev and Harmuth 116 (2003), Feng et al. (2013), Li et al. (2014)). Saldi and Wen (2016) adopted a 117 specific model for the failure assessment of hydrogen cylinders for automotive 118 applications. In the review presented by (Godoy (2016)), the buckling prob-119 lems of atmospheric tanks under static or quasi-static loads were investigated 120 and specific modeling approaches were discussed considering accidental fire 121

exposure. The coupled assessment of the thermal and mechanical response 122 was undertaken for light fuel oil storages (Rebec et al. (2016)) and pressurized 123 gas pipelines (Jang et al. (2015)). In this case, FEM and CFD are adopted 124 to reproduce heat flux exposure conditions and to predict the eventual fail-125 ure conditions. To the best of our knowledge, this was not undertaken in a 126 coupled way for pressurized tanks. In fact, Landucci et al. (Landucci et al. 127 (2009a), Landucci et al. (2009b), Landucci et al. (2009c)) and Manu et al. 128 (2009) provided detailed examples of the simulation of LPG tanks exposed to 129 fire, in order to estimate the time to failure and to characterize the escalation 130 scenarios. However, in this latter case, the integration of different modeling 131 approaches for the comprehensive characterization of cascading event chains 132 is not yet consolidated. 133

The present study focuses on the analysis of pressurized vessels exposed to fire. This type of accidental situation may lead to severe cascading events following the catastrophic rupture of vessels. In the case of storage or processing of flammable liquefied gases under pressure, such as propane, butane, propylene, etc., a BLEVE (Boiling Liquid Expanding Vapour Explosion) may occur (Reid (1979), Venart (1999)), eventually followed by fireball (Abbasi and Abbasi (2007), Maillette and Birk (1996)).

A multi-level approach for the advanced simulation of accident scenarios involving cascading events will be proposed. This is based on coupling advanced boundary condition, based on integral modeling, to distributed parameters modeling. In particular, the work aims at improving a previous CFD model of a pressurised tank described in D'Aulisa et al. (2014) and Landucci et al. (2016a) in order to assess its response in case of complex

fire exposure conditions. The latter are imposed by simulating the pri-147 mary fire through integral models available in literature (Mannan (2012), 148 Van Den Bosh and Weterings (2005)) and coupling the results into the CFD 149 model through bespoke subroutines. The potentiality of the novel approach 150 described in Section 2 will be tested through the application to a large-scale 151 case study defined in Section 3, highlighting the computational requirements 152 and main novelties of the present work in Section 4. Results are shown and 153 discussed in Section 5. 154

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¹⁵⁶ 2. Methodology

The present study focuses on the analysis of cascading events triggered 157 by fire. The sketch of the problem in shown in Figure 1. A pressurized vessel 158 exposed to a pool fire receives heat due to radiation and convection, and 159 subsequently heat is transferred by conduction through the vessel wall to the 160 interior, leading to an increase of vapor and liquid temperature and pressure, 161 as described by Moodie (1988). Significant heat dissipation occurs in the liq-162 uid with respect to the vapor due to the higher heat transfer coefficient of 163 the liquid phase, which may be one or two orders of magnitude higher than 164 that of the vapor (Aydemir et al. (1988), Birk (1989), Moodie (1988)). 165

The heat-up of the liquid leads to strong recirculation phenomena, which cause an upward flow of the hot liquid in the boundary layer and a downward flow in the central region of the tank (Birk and Cunningham (1996)). Consequently, a buoyancy-driven flow is induced by density variations, so that a vertical temperature gradient is established inside the tank, i.e. the liquid is thermally stratified (see for instance Birk and Cunningham (1996),
Shi et al. (2013), D'Aulisa et al. (2014) and references therein). Hence the
vapor at the interface is saturated at the temperature of the warmest liquid
layer.

In case of non-uniform exposure of the tank to a fire, it may happen that some regions of the vessel receive more heat load than others. This is schematically shown in Figure 1 where in this case the vessel is subjected to high heat load from the right, whereas to small or nearly zero load from the left.

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[Figure 1 about here.]

The present work aims at evaluating the non-uniform heat load on a pressurized tank, generated from a distant radiation source (i.e. a pool fire), and at analyzing the effect of such non-uniform load distribution on the vessel response.

In theory a full simulation of the problem would require the modeling of a 185 3-dimensional pressurized vessel containing a multi-phase flow and exposed 186 to a pool fire, hence to a transient turbulent reactive flow. This would lead to 187 computationally unfeasible simulations because of the large number of equa-188 tions needed to describe all phenomena (turbulence, reaction, mass and heat 189 inter-phase transfer and radiation), the large number of cells required for a 190 3-dimensional geometry including both fluid and solid domains as well as the 191 time discretization required for the transient feature of the problem. Indeed 192 to our knowledge all numerical investigations concentrate on either the fire 193 simulation, thus neglecting the behavior of the lading fluid in the target ves-194 sel (e.g. Masum Jujuly et al. (2015)), or the multiphase flow inside the tank 195

exposed to fire, thus simplifying the treatment of heat exposure conditions (e.g., Bi et al. (2011), D'Aulisa et al. (2014), Landucci et al. (2016a)). Hence, in the present work the problem is decoupled by addressing separately the pool fire modeling and its effect on the pressurized tank. The underlying assumption is that the pool fire is characterized by a timescale much larger than the storage tank dynamics, so that the flame is considered to be at steady state, whereas transient simulations are adopted for the tank.

Moreover, since the study is not focused on the pool fire itself but rather on its impact on the target tank, the pool fire is modeled using an integral approach. The idea is indeed similar to that used by Pontiggia et al. (2011) to analyze a major accident from a LPG rail-car rupture in an urban area; in this case an integral model was used to evaluate the LPG release that was incorporated in the CFD dispersion model in the urban area as a source term.

210 3. Test Case

Figure 2 shows the layout considered for the analysis of the case study. 211 In particular, the tank farm of a refinery is adopted as reference installa-212 tion. The tank farm is constituted by atmospheric and pressurized tanks. 213 In particular, T1 is an atmospheric tank, storing crude oil (assimilated as 214 n-hexane) and T2 is a pressurized vessel storing LPG (assimilated as pure 215 propane). The main features of the tanks are summarized in Table 1. It was 216 then assumed that a failure in T1 leads to a pool fire in the tank catch basin; 217 the pool fire radiation affects T2 which is located about 20 m far from the 218 catch basin edge (see Figure 2). 219

[Table 1 about here.]

The consequences of the pool fire in T1 catch basin were evaluated using conventional literature integral models based on surface emissive power approximation, as described by Mannan (2012) and Van Den Bosh and Weterings (2005). A single set of meteorological parameters was used to calculate the consequences of the pool fire, in particular:

• wind velocity =
$$5 \text{ m/s}$$

• atmospheric neutral conditions (stability class D)

• relative humidity =
$$50\%$$

• ambient temperature =
$$20 \, ^{\circ}\mathrm{C}$$

A uniform wind direction was assumed for the sake of simplicity (see Figure 231 2), the flame was considered to be stable and in steady state conditions. 232 More details on the calculation procedure for the pool fire consequences are 233 summarized in Appendix A. Pool fire radiation simulation allowed gathering 234 non-uniform boundary conditions for the analysis of the heat-up of target 235 T2 through the CFD model, thus providing an example of coupling different 236 kind of models. For the sake of comparison, the same vessel was simulated 237 assuming a uniform incoming heat radiation distribution, as carried out in 238 conventional literature approaches (D'Aulisa et al. (2014), Landucci et al. 230 (2016a)).240

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241 4. CFD Model

242 4.1. Computational Domain and Grid

Since the storage tank has a length much larger, i.e. more than six times. 243 than its diameter, a 2-dimensional (2D) domain corresponding to a cross 244 section of the T2 tank was chosen. This approach was aimed at reducing 245 computational efforts even though some end effects may alter the boundary 246 layer and the warm top layer of the liquid, affecting the pressurization rate. 247 The grid was generated with the O-grid method using the ICEM software, 248 by ANSYS Inc. and, hence, it is block structured. The grid is uniform over 249 all the domain, except near the wall where a refinement was applied to better 250 capture velocity profiles. 251

The number of cells is 268k and it is extremely large considered the sim-252 plicity and 2D feature of the domain; however such a fine grid was found to 253 be necessary to capture the liquid level rise due to the temperature increase 254 in the storage tank. This is one of the main improvements with respect to 255 other works in literature (Bi et al. (2011), D'Aulisa et al. (2014), Landucci 256 et al. (2016a)), where the grid is refined near the liquid-vapor interface, that 257 is known a priori, in order avoid any convergence problems due to the evapo-258 ration/condensation phenomena. In other approaches, the domain is divided 259 into two sub-domains, one for the liquid and one for the vapor phase, as 260 done by D'Aulisa et al. (2014). In such a manner the grid can be coarse, 261 with a significant saving of CPU time; however the change of liquid level 262 due heat-up cannot be predicted effectively. Logically, also in this case the 263 computational grid is suited only for a given initial liquid level. 264

²⁶⁵ Instead, the approach of the present work aims at capturing the liquid level

rise for any initial filling level, through the adoption of the same computational grid. For sake of brevity, results are shown just for a single filling level.

269 4.2. Physical Model

The physical model was based on the Volume of Fluid (VOF) approach that enables the prediction of multi-phase flows in which the interfaces are clearly identified (Hirt and Nichols (1981))

The model assumes that the each control volume contains just one phase or the interface between the phases. This is determined by the volume fraction α_L of, say, the liquid phase, identifying three cases:

- if $\alpha_L = 0$ the cell is completely full of vapor;
- if $\alpha_L = 1$ the cell is completely full of liquid;
- if $0 < \alpha_L < 1$ the cell contains the vapor-liquid interface.

In presence of a turbulent flow, the governing equations that are solved inthe domain are:

• continuity equation

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{U}) = 0 \tag{1}$$

• momentum equation

$$\frac{\partial \left(\rho \mathbf{U}\right)}{\partial t} + \nabla \cdot \left(\rho \mathbf{U}\mathbf{U}\right) = -\nabla P + \nabla \cdot \left(\mu + \mu_T\right) \left(\nabla \mathbf{U} + \nabla \mathbf{U}^T\right) + \mathbf{F} \quad (2)$$

• energy equation

$$\frac{\partial \left(\rho c_p T\right)}{\partial t} + \nabla \cdot \left[\mathbf{U}\left(\rho c_p T + P\right)\right] = \nabla \cdot \left[\left(\kappa + \frac{c_p \mu_T}{P r_T}\right) \nabla T\right] + S_h \quad (3)$$

where **U**, *T* and *P* are the mean velocity vector, temperature and pressure, respectively, and the superscript *T* indicates the transpose of a vector. μ_T and Pr_T are the turbulent viscosity and Prandtl number, respectively. The former is determined through the standard κ_{ε} turbulence model with scalable wall functions, whereas the $Pr_T = 0.85$ (Tu et al. (2013)).

The properties appearing in the transport equations are determined by the presence of the component phases in each control volume. For instance, density, specific heat and thermal conductivity are computed by the following expressions:

$$\rho = \alpha_L \rho_L + (1 - \alpha_L) \rho_V \tag{4}$$

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$$c_p = \alpha_L c_{pL} + (1 - \alpha_L) c_{pV} \tag{5}$$

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$$\kappa = \alpha_L \kappa_L + (1 - \alpha_L) \kappa_V \tag{6}$$

The liquid was modeled as incompressible; even though its density was allowed to vary with temperature in the body force term \mathbf{F} of the momentum equation using the Boussinesq model:

$$\mathbf{F} = (\rho - \rho_0) \,\mathbf{g} \approx -\rho_0 \beta_T \left(T - T_0\right) \mathbf{g} \tag{7}$$

where ρ_0 is the constant density of the fluid, β_T is the thermal expansion and T_0 is the operating temperature. No momentum exchange between the liquid and the vapor phase due to surface tension σ is considered because it is less important than the gravitational body force, i.e. the Eotvos number $Eo = \frac{(\rho_L - \rho_v)gL^2}{\sigma} \gg 1$, where L is the characteristic length. Therefore, the interface between liquid and vapor can be considered waveless. The evolution of the vapor-liquid interface was tracked by solving a a volume fraction continuity equation for each phase except for the primary phase. In this case, setting the vapor phase as a primary phase, the volume fraction continuity equation is solved only for the secondary phase, i.e. the liquid phase. All other equations (momentum, energy, radiation) are shared by the phases. For the liquid phase, the volume fraction continuity equation is:

$$\frac{\partial \left(\alpha_L \rho_L\right)}{\partial t} + \nabla \cdot \left(\alpha_L \rho_L \mathbf{U}\right) = S_{\alpha_L} + \left(\dot{m}_{VL} - \dot{m}_{LV}\right) \tag{8}$$

while for the vapor phase, the volume fraction in each cell is computed following the mathematical constraint:

$$\alpha_L + \alpha_V = 1 \tag{9}$$

in each cell. In Equation 8 S_{α_L} represents the rate of increase of liquid 312 volume fraction due to external liquid mass source term (that is zero in the 313 present closed case), whereas $\dot{m}_{VL} - \dot{m}_{LV}$ is the rate of increase of liquid mass 314 due to the difference between the mass transfer from vapor to liquid phase 315 minus the mass transfer from liquid to vapor phase. To simulate in detail 316 the evaporation/condensation phenomenon, hence to determine \dot{m}_{VL} and/or 317 \dot{m}_{LV} the "Lee Model" was adopted (Lee (1980)). The model assumes that 318 the mass is transferred at constant pressure and at quasi thermo-equilibrium 319 state, so that the mass transfer can be estimated (for evaporation, ie for 320 $T > T_{sat}$ as: 321

$$\dot{m}_{LV} - \dot{m}_{VL} = r_i \alpha_L \rho_L \frac{T - T_{sat}}{T_{sat}} \tag{10}$$

where T_{sat} is the saturation temperature at the given pressure, and r_i in the mass transfer intensity factor that was taken $r_i = 0.1 \text{ s}^{-1}$ as suggested by De Schepper et al. (2009) for the simulation of boiling from hydrocarbon feedstock.

For what concern the energy balance and the estimation of the heat transfer during the evaporation or condensation process, only one expression is required, in which the energy source terms related to evaporation $(S_{h,evap}h)$ and condensation $(S_{h,cond}h)$ are expressed through the latent heat of vaporization λ_0 , for instance:

$$S_{h,evap} = \dot{m}_{LV} \lambda_0 \tag{11}$$

Radiation was modeled through the Surface to Surface (S2S) Model, that 331 accounts for the radiation exchange in an enclosure of gray-diffuse surfaces 332 through view factors and neglects any absorption, emission, or scattering. 333 Hence the fluid was considered to not participate to radiation; this is moti-334 vated by the low vapor temperature (lower than 400K). Moreover the S2S 335 model is computationally less demanding than other models such as the Dis-336 crete Order, P-n approximation, Discrete Transfer and Monte Carlo method, 337 that involve the calculation of the interaction with the participating medium 338 and ray or photon tracing techniques (Modest (2003)). 339

340 4.3. Physical Properties

The LPG stored in the vessels exposed to the fire is a mixture of propane and butane, with high propane mass fraction (i.e. 95-98%). Hence, in the CFD model the LPG is assumed to be pure propane, thus neglecting the presence of heavier components. The saturation temperature T_{sat} (K) and the latent heat of vaporization λ_0 (J kg⁻¹) are expressed as a function of the absolute pressure P (Pa), which changes with time, through polynomial ³⁴⁷ relationship as made by D'Aulisa et al. (2014).

$$\lambda_0 = 0.0682P + 403262 \tag{12}$$

348 and

$$T_{sat} = -6.0 \cdot 10^{-12} P^2 + 5.0 \cdot 10^{-5} P + 253.76 \tag{13}$$

The liquid and vapor properties are implemented as a polynomial or power law function of the absolute temperature (D'Aulisa et al. (2014)) from available thermodynamic data (Green and Perry (2008)). The simplified correlations are shown in Table 2.

Since the fluid in storage condition and during the exposure to fire is at considerable pressure (more than 10 bar), the Peng Robinson equation of state (PR-EOS) was used to estimate the vapor density:

$$P = \frac{RT}{v_m - b_{PR}} - \frac{a_{PR}\psi(T)}{v_m^2 + 2b_{PR}v_m - b_{PR}^2}$$
(14)

 $_{357}$ where P is expressed in bar. In the above expression:

• $R = 83.144 \text{ cm}^3 \text{ bar} / \text{mol K};$

•
$$a_{PR} = 0.45724 \frac{R^2 T_c^2}{P_C} \text{ cm}^6 \text{ bar/mol}^2;$$

so
$$b_{PR} = 0.0778 \frac{RT_C}{P_C} \text{ cm}^3/\text{mol};$$

•
$$\psi(T) = 1 + (0.37464 + 1.54226\omega - 0.26992\omega^2) \left[1 - \left(\frac{T}{T_C}\right)^2\right]$$

The critical temperature and pressure for propane are $T_C = 369.9$ K and $P_C = 42.051$ bar, respectively, whereas the acentric factor $\omega = 0.152$. A detailed validation of the physical model is reported in the work of (D'Aulisa et al. (2014)).

367 4.4. Integral Method for Heat Radiation

In order to estimate the heat flux conditions affecting the tank exposed 368 to fire, an integral model for pool fire radiation simulation was adopted. This 369 allowed determining the heat flux conditions summarized in Section 4.5. The 370 procedure for the consequence assessment of pool fire radiation through inte-371 gral models is well known in the literature and extended details are reported 372 elsewhere (Mannan (2012), Van Den Bosh and Weterings (2005)). Figure 373 3 summarizes the procedure adopted in the present study for simulation of 374 pool fire radiation and the main equations involved. For the sake of brevity, 375 more details on the calculation procedure are discussed in Appendix A. 376

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[Figure 3 about here.]

378 4.5. Boundary Conditions

- Two different types of boundary conditions were applied to the tank walls.
- non-uniform heat flux, corresponding to the incident radiation evaluated with the integral model for pool fire, as described in Appendix A. The variation of incident flux with the angular coordinate of the tank wall is shown in Figure 4. It can be observed a maximum radiation of 77 kw/m² at 45°. Between 90° and 150° still some radiation exists, whereas negligible incident heat flux is between 150° and 270°;

• uniform heat flux of 26.2 kW/m^2 was applied at the walls. Such value 386 was obtained by averaging the heat radiation distribution predicted by the pool fire model. 388

The former boundary condition was set through a C++ subroutines described 380 in Appendix A. It is worth noting that, since the boundary conditions con-390 sisted of heat flux value, the wall thickness was not specified. However, the 391 approach may be easily extended by adding solid domains for the walls in 392 case an accurate estimation of temperature profiles inside the walls is re-393 quired, as for instance for the analysis of fireproofing performance (Landucci 394 et al. (2009b)). 395

[Figure 4 about here.]

4.6. Solver 397

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A pressure based solver with an implicit time advancement, available in 398 Fluent v. 16, by Ansys Inc., was employed. The time step was chosen in order 399 to ensure a Courant number lower than 5. A first order upwind discretization 400 scheme was used for all equations and the SIMPLE algorithm was applied for 401 the pressure-velocity coupling. Normalized residuals for all equations were 402 typically well below 10^{-6} . One hour of CPU time was needed to cover 1 s of 403 real time when run on 32 threads. Simulations were run to cover the time up 404 to tank pressurization corresponding to the set pressure of the release valve 405 (see Table 1). Hence, a single simulation took more than 1 week. 406

5. Results and Discussion 407

Figure 5 illustrates the distribution of temperature in the liquid phase 408 at different times predicted using an uniform heat flux distribution at the 409

walls. It can be observed a mirror-symmetric pattern with respect to the vertical tank mid-plane. The high temperature region evolves from the walls (see the snapshot at time $\tau = 5$ s) towards the liquid surface and then it enters ($\tau = 15$ s) in the middle to extend downwards. Subsequently, the low temperature region progressively moves to the bottom, leading to a thermal stratification. The liquid level rises of approximately 0.1 m in 292s.

The motion originated from the temperature gradients is depicted at the 416 same time steps in Figure 6. At $\tau = 5$ s the high velocity regions are located 417 near the wall, where an upward motion is established due to buoyancy effects. 418 It is worthy to remind that no slip velocity is set to the wall; however, the 419 boundary layer thickness is so small (i.e., less than a few millimeters) that the 420 region in which velocity goes from the bulk value to zero is not discernible. 421 The magnitude of the convective velocity V_c can be roughly estimated by 422 balancing inertial and buoyancy forces, so that the Grashof number, Gr, can 423 be interpreted as the square of the Reynolds number, Re (Mauri (2015)). 424

$$Gr = \frac{L^3 g \beta_T \Delta T}{\nu^2} = \left(\frac{V_c L}{\nu}\right)^2 = Re^2 \tag{15}$$

where ΔT is the temperature difference driving natural convection, L is the characteristic length and ν is the kinematic viscosity. Hence V_c is proportional to \sqrt{Gr} and hence to $\sqrt{\Delta T}$.

Small vortical structures are observed near the liquid-vapor interface very close the walls. Then, such structures increase and move towards the vessel mid-plane, as shown at $\tau = 15$ s when a pair of counter-rotating recirculation regions is well evident. These promote the motion of the fluid near the vessel mid-plane towards the bottom, with a velocity of about 0.3 m/s. The flow is fully turbulent; since the viscosity of propane is low (around 10^{-4} Pa s) the resulting Re is above 10^5 by taking the average velocity across the liquid phase (i.e. 0.08 m/s at $\tau = 15 \text{ s}$) and the characteristic length equal to 1/10of the tank diameter. The induced turbulent fluctuations, that can be estimated from the turbulent kinetic energy, can result in a turbulent intensity above 10% in some regions fo the tank.

Subsequently, the strength of the vortical structures progressively diminishes due to mixing that smooths temperature gradients, thus reducing the Grashof number and hence V_c , finally leading to the thermal stratification depicted in Figure 5f.

In Figure 6c the flow appears slightly asymmetric in the low velocity region near the vessel bottom. In fact, despite the geometry and boundary conditions are symmetric, the flow can be asymmetric due to the establishment of vortical structures, that are more likely promoted in the present unsteady conditions.

- 448 [Figure 5 about here.]
 - [Figure 6 about here.]

449

Similarly, Figure 7 shows the contours of temperature in the liquid phase at different times, as evaluated using the non-uniform heat flux distribution predicted through the integral model, described in Appendix A. For sake of comparison, the sampling times are the same as those used in Figure 5, except for the last that refers to opening of the release valve.

It clearly appears the asymmetric feature of the distribution, with the high temperature region originating from the right side (exposed to the fire) and spreading on the top (see $\tau = 15$ s), then moving towards the bottom on the left side (not exposed to the fire). This is confirmed in Figure 8 that shows,
for the same time, a single vortical structure that promotes a descending
motion on the left side of the tank.

Subsequently, thermal stratification can be also observed, however with dif-461 ferent features. In fact at the same time, the bulk temperature estimated 462 with uniform heat flux conditions is higher than that obtained with the non-463 uniform ones. In the uniform case the fluid motion is very effective as it 464 comes from both sides on the tank; this promotes the overall heat-up of the 465 lading. Conversely, in the non-uniform case, the fluid motion is unable to 466 affect all the lading as it comes from just one side of the tank; hence the 467 heat-up process is slower than for the uniform case (compare Figure 5f and 468 7f). 469

After $\tau = 200$ s the asymmetry of the temperature distribution in the nonuniform heat flux is less visible with respect to the initial times due to the weaker motion induced by the lower Gr.

It is worthy to notice that since the final time is imposed as the one corresponding to the safety valve opening, the temperature at the liquid-vapor interface, which drives the pressure (see Section 2), is the same for the two cases. However, due to the reduced recirculation observed for the nonuniform heat flux case, the upper liquid layer tends to heat-up faster than for the uniform case.

479

480

[Figure 7 about here.]

481

[Figure 8 about here.]

Subsequently the pressure build up is quicker in the non-uniform heat 482 flux case as also reported in Figure 9a. Such pressure represents an average 483 over the vapor phase, even though the observed differences between different 484 locations in the vapor were less than 0.02%. The release valve pressure 485 opening is predicted after 230 s for the non-uniform heat flux case, and after 486 292 s for the uniform one. Therefore using a more sophisticated approach, 487 despite the large computational and setup efforts, leads to results that are 488 significantly different from the simple uniform heat flux impacting on the 489 tank. In particular, this latter assumption leads to an overestimation of 490 about 30% of the pressurization time, thus leading to a less conservative 491 prediction. Figure 9b shows the evaporation rate, evaluated from the time 492 derivative of the liquid mass, as a function of time for both uniform and 493 non-uniform heat flux. The evaporation rate is similar up to approximately 494 $\tau = 100$; subsequently it is larger for uniform than for non-uniform heat flux. 495 This may be imputed to the dependence of latent heat on pressure, reported 496 in Equation 12. Such latent heat is evaluated as a function of time for both 497 cases in Figure 9; after $\tau \approx 100$ s the latent heat for non-uniform heat flux 498 is higher than for uniform heat flux, so that less LPG evaporates for a given 499 heat flux. 500

501

[Figure 9 about here.]

502 6. Conclusions

⁵⁰³ Safety enhancement of chemical and process plants asks for innovative ⁵⁰⁴ tools in order to support QRA studies. In particular, in order to capture the transient and dynamic nature of complex accident scenarios such as cascading events triggered by fire, specific methods are needed to obtain accurate predictions. The present study coupled integral and distributed parameters models to simulate cascading events triggered by fire. In particular, CFD modeling of a pressurized vessel exposed to fire was carried out by imposing heat flux conditions at the walls derived from an integral model for pool fire radiation simulation.

The application to a case study of industrial interest allowed obtaining results 512 that are hardly derivable with simplified models and assumptions and which 513 may be interpreted in a dual perspective. In fact, the simulation of tanks ex-514 posed to realistic heat source types allowed determining the influence of the 515 induced buoyancy driven flow on the pressurization rate, thus supporting the 516 investigation of complex stratification and recirculation phenomena. Then, 517 the results showed the influence of realistic fire scenarios on the dynamic 518 evolution of the heat-up of potential target vessels, thus gathering key in-510 formation about the possible timing for the deployment of emergency teams 520 and resources. 521

Moreover, the present simulation approach may be extended to vessels con-522 taining different types of substances, featuring different operative conditions 523 and geometries. This may allow to gather an extended data set of vessels 524 response during fire exposure, thus supporting the development of vulnera-525 bility models for process equipment exposed to fire, such as probit functions 526 (e.g., see Landucci et al. (2009a) for more details). Finally, it is worth men-527 tioning that the computational and setup efforts make unfeasible to extend 528 the approach to all plant sections, so that the most critical ones should be 529

⁵³⁰ previously selected through screening criteria (Cozzani et al. (2007)).

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781 List of Figures

782	1	Conceptual scheme of the problem and methodology	38
783	2	Overview of the industrial facility considered for the case study.	39
784	3	Flow chart illustrating the model fore determining the pool	
785		fire radiation.	40
786	4	Incident radiation in kW/m^2 on the tank at different angles.	
787		Fire comes from the right, see Figure 1	41
788	5	Dynamic distribution of temperature (K) in the liquid phase	
789		under uniform heat flux.	42
790	6	Dynamic flow field in the liquid phase under uniform heat flux.	
791		Vectors are colored by axial velocity (m/s)	43
792	7	Dynamic distribution of temperature (K) in the liquid phase	
793		under non-uniform heat flux	44
794	8	Dynamic flow field in the liquid phase under non-uniform heat	
795		flux. Vectors are colored by axial velocity (m/s)	45
796	9	Pressure (a) and evaporation rate (b) as a function of time	
797		predicted with uniform and non-uniform heat flux boundary	
798		conditions	46
799	A.10	Scheme of a titled pool fire	57
800	A.11	Scheme adopted for view factor determination	58
801	A.12	Section of the subroutine specifying non-uniform boundary	
802		conditions	59



Figure 1: Conceptual scheme of the problem and methodology



Figure 2: Overview of the industrial facility considered for the case study.



Figure 3: Flow chart illustrating the model fore determining the pool fire radiation.



Figure 4: Incident radiation in $\rm kW/m^2$ on the tank at different angles. Fire comes from the right, see Figure 1.



Figure 5: Dynamic distribution of temperature (K) in the liquid phase under uniform heat flux.



Figure 6: Dynamic flow field in the liquid phase under uniform heat flux. Vectors are colored by axial velocity (m/s).



Figure 7: Dynamic distribution of temperature (K) in the liquid phase under non-uniform heat flux.



Figure 8: Dynamic flow field in the liquid phase under non-uniform heat flux. Vectors are colored by axial velocity (m/s).



Figure 9: Pressure (a) and evaporation rate (b) as a function of time predicted with uniform and non-uniform heat flux boundary conditions.

803 List of Tables

804	1	Main features of the tanks considered for the case study	48
805	2	Correlations used in the CFD model to evaluate propane prop-	
806		erties	49
807	A.3	Physical properties of n-hexane	61

Property	Tank T1	Tank T2
Nominal diameter (m)	42	3.2
Nominal height/length (m)	5.4	19.4
Maximum wall thickness (mm)	12.5	27
Design pressure $(barg)^*$	0.02	17
Nominal volume (m^3)	7500	150
Stored fluid	n-hexane	propane
Filling ratio (-)	0.7	0.9
Inventory (t)	3439	78
Area of the catch basin (m^2)	3575	18000**

Table 1: Main features of the tanks considered for the case study.

*assumed as the release valve set pressure **pressurized tanks share the same catch basin area

Property	Units	Correlation
liquid density ρ_L	$ m kg/m^3$	$\rho_L = -24.063 + 4.9636T - 0.0109T^2$
vapour density ρ_V	$ m kg/m^3$	PengRobinson EOS
liquid heat capacity $c_{p,L}$	J/(kg K)	$c_{p,L} = 36309 - 230.2T + 0.3941T^2$
vapour heat capacity $c_{p,V}$	$\rm J/(kgK)$	$c_{p,V} = 345.58 + 4.4019T$
liquid thermal conductivity κ_L	W/(mK)	$\kappa_L = 0.26755 - 6.6 \cdot 10^{-4}T + 2.77 \cdot 10^{-7}T^2$
vapour thermal conductivity κ_V	W/(mK)	$\kappa_V = -0.0088 + 6.0 \cdot 10^{-5}T + 1.0 \cdot 10^{-7}T^2$
liquid dynamic viscosity μ_L	Pa s	$\mu_L = 709137 T^{-3.986}$
vapour dynamic viscosity μ_V	Pa s	$\mu_V = 4.9054 \cdot 10^{-8} T^{0.90125}$

Table 2: Correlations used in the CFD model to evaluate propane properties.

Appendix A. Procedure for the evaluation of pool fire heat radiation effects

The procedure for the consequence assessment of pool fire radiation is well 810 known in the literature and is summarized in the following. In this work, 811 the application of the procedure allowed obtaining the boundary conditions 812 for the CFD simulation described in Section 5. For the analysis of the case 813 study, a crude oil pool fire (assimilated as pure n-hexane) was simulated. 814 The physical properties of n-hexane are reported in Table A.3. More details 815 on integral models adopted for pool fire simulation are extensively reported 816 elsewhere Mannan (2012) Van Den Bosh and Weterings (2005). 817

⁸¹⁹ Appendix A.1. Determination of pool diameter

818

The first step is aimed at determining the liquid pool equivalent diameter (D_p) , since the liquid hydrocarbon from tank T1 is spilled into a rectangular catch basin, covering its entire surface (see Figure 2). The following relationship is adopted:

$$D_p = \sqrt{\frac{4}{\pi A_p}} \tag{A.1}$$

where A_p is the area of the catch basin (see Table 1).

⁸²⁵ Appendix A.2. Evaluation of the burning rate

The burning rate $(m'', \text{ in kg s}^{-1} \text{ m}^{-2})$ is defined as the rate of evaporation of material per unit surface on the pool. For large pool fires (e.g., $D_p > 1 \text{ m})$, m'' depends only on the type of substance and may be evaluated as follows:

$$m'' = \frac{0.001 \cdot H_C}{H_V + c_p \left(T^0 - T_{atm}\right)}$$
(A.2)

where H_C and H_V (in J/kg) are respectively the heat of combustion and of vaporization of the substance at the pool temperature T^0 (see Table A.3); c_p is the average liquid heat capacity and T_{atm} is the ambient temperature. It is worth mentioning that, for an evaporating pool, such as in the present case, pool temperature is equal to the atmospheric temperature (hence, $T^0 - T_{atm} =$ 0).

⁸³⁵ Appendix A.3. Evaluation of flame geometry

In the so called solid flame approach, the flame is simulated as a solid of a given geometry featuring an average emissivity. In the present work a tilted cylindrical shape was determined for the flame, considering the burning rate and the effect of wind on the flame structure (see Figure A.10).

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Firstly, the scaled wind velocity u^* is evaluated as follows:

$$u^* = \frac{u_w}{\left(\frac{gm''D_p}{\rho_a}\right)^{0.33}} \tag{A.3}$$

where u_w is the wind velocity at a height of 10 m, ρ_a is the air density (= 1.25 kg m⁻³) and g is the gravitational acceleration (= 9.81 m s⁻²). The scaled velocity allows accounting for wind tilting effects in the pool fire, which geometrical parameters were estimated adopting the following semi-empirical correlation:

$$\frac{H_f}{D_f} = 55 \left(\frac{m''}{\rho_a \sqrt{gD_p}}\right)^{0.67} (u^*)^{-0.21}$$
(A.4)

where H_f (m) is the flame height and D_f is the flame diameter. The flame side length h_f (see Figure A.10) is then determined as a function of the flame height and scaled wind velocity:

$$h_f = \frac{H_f}{\cos(\theta)} = H_f \sqrt{u^*} \tag{A.5}$$

where θ is the tilted angle (see Figure A.10). Finally, the flame diameter (D_f , in m) was obtained applying the following relationship, accounting for the displacement due to wind:

$$\frac{D_f}{D_p} = a_p \left(Fr_{10}\right)^{b_p} \tag{A.6}$$

in which Fr_{10} is the Froude number at 10 m defined as follows:

$$Fr_{10} = \frac{u_w^2}{gD_p} \tag{A.7}$$

The coefficients a_p and b_p both depend on the flame geometry type; for cylindrically shaped flames $a_p = 1.5$ and $b_p = 0.069$ Van Den Bosh and Weterings (2005)

⁸⁵⁷ Appendix A.4. Evaluation of surface emissive power (SEP)

Once having determined the flame shape, the surface emissive power (SEP) can be estimated. SEP indicates the heat radiated outwards per unit surface are of the flame. The following correlation was adopted to determine the maximum value of SEP, without accounting for the effect of soot, thus obtaining a conservative evaluation Van Den Bosh and Weterings (2005):

$$SEP = \frac{F_s m'' H_C}{1 + 4\frac{h_f}{D_f}} \tag{A.8}$$

in which F_s indicates radiation fraction (the amount of heat generated by the flame which is transferred by radiation).

⁸⁶⁶ Appendix A.5. Evaluation of atmospheric transmissivity

The atmospheric transmissivity (τ_a) accounts for the fact that the emitted 867 radiation is partly absorbed by the air present between the flame and the 868 target receiver. The transmissivity depends on the absorbing properties of 869 the components of the ambient air in relationship to the emission spectrum 870 of the fire. Neglecting the presence of carbon dioxide in the atmospheric 871 air, water vapor was considered as the main absorbing component within 872 the wave length area of the heat radiation, thus the following approximating 873 expression was adopted (Mannan (2012)): 874

$$\tau_a = c_w \left(X P_w \right)^{-0.09} \tag{A.9}$$

in which c_w is a constant (= 2.02 (N/m)^{0.09}) and the following conditions is verified:

$$10^4 < XP_w < 10^6 \text{N/m}$$
 (A.10)

where X is the distance of the receiver (see Figure A.10) and P_w is the partial pressure of water vapour in the atmospheric air (thus, function of the relative humidity).

⁸⁸⁰ Appendix A.6. Evaluation of the geometrical view factor

The geometrical view factor (F_v) is the ratio between the received and the emitted radiation power per unit surface. The factor is determined by the flame dimensions and shape, and by the relative position and orientation of the receiving object. Considering the representation reported in Figure
A.11, the geometrical view factor is defined as follows:

$$F_{v_{dA1,dA2}} = \frac{1}{\pi} \iint \left(\frac{\cos(\beta_1)\cos(\beta_2)}{X^2} \right) dA_2$$
 (A.11)

where X is the distance between the centers of dA_1 and dA_2 , β_1 is the angle of the normal vector to plane dA_1 and the line connecting dA_1 and dA_2 and β_2 is the angle of the vector to plane dA_2 and the line connecting dA_1 and dA_2 .

Typically, simple flame shapes are taken for the calculations such as 891 sphere, cylinder and flat plate. In the present study, the view factor of 892 a cylinder may be used. The approach developed by Raj Raj (2005) was 893 adopted in the present study to estimate the view factor on the target tank. 894 The reader is referred to Raj (2005) for more details on the procedure. Prac-895 tically the tank surface was divided into 16 sectors and, for each of them, the 896 view factor F_{v_i} from the cylinder (i.e., the pool fire) to the sector centroid 897 was estimated. 898

⁸⁹⁹ Appendix A.7. Evaluation of heat flux on the receiver

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Finally, the heat flux q_i (W m⁻²) from the pool fire on the tank surface i-th sector is evaluated through the radiative heat transfer equation Modest (2003)

$$q_i = SEP \cdot F_{v_i} \cdot \tau_{a_i} \tag{A.12}$$

The procedure described above allowed obtaining the incoming heat flux 903 distribution on the external surface of the tank, at 16 discrete locations. 904 Subsequently the heat flux was linearized between such locations, resulting 905 in the heat flux distribution shown in Figure 4. Such distribution was set 906 as boundary condition for the CFD model through a C++ User Defined 907 Functions that basically checks the (x, y) coordinates of the location in the 908 boundary to determine the angular coordinate and thus associate the corre-909 sponding heat flux. A section of the subroutine is reported in Figure A.12 910

[Figure 12 about here.]

911

55

912 List of Figures



Figure A.10: Scheme of a titled pool fire



Figure A.11: Scheme adopted for view factor determination

```
#include "udf.h"
DEFINE_PROFILE(flux_heat_space, thread, position)
{
    face_t f;
    double pos[ND_ND];
    double x,y; /*coordinates*/
    double a;
                 /*tank radius in m*/
    double PI=3.14159;
    begin_f_loop(f, thread)
    {
        F_CENTROID(pos, f, thread);
        x = pos[0];
        y = pos[1];
        a = 1.6;
        if (x <= -cos(PI/6)*a && y <= 0.)
        {F_PROFILE(f, thread, position) = 0.;}
        else if (x <= -cos(PI/6)*a && y > 0.)
        {F_PROFILE(f, thread, position) = 4625*y;}
        else if (x \le -\cos(PI/4) \ge \delta x > -\cos(PI/6) \ge \delta y \le 0.)
        {F_PROFILE(f, thread, position) = 0.;}
        else if (x <= -cos(PI/4)*a && x > -cos(PI/6)*a && y > 0.)
        {F_PROFILE(f, thread, position) = 21125.06*y -13200.05;}
        else if ....
        ----
        else if (x > cos(PI/6)*a && y <= 0.)
        {F_PROFILE(f, thread, position) = 36475*y + 57900;}
        else
        {F_PROFILE(f, thread, position) = 22500*y + 57900;}
    }
    end_f_loop(f,thread)
}
```

Figure A.12: Section of the subroutine specifying non-uniform boundary conditions.

913 List of Tables

Property	Units	Value	Reference		
Liquid density ρ_L	${ m kg}~{ m m}^{-3}$	655	Liley et al. (1999)		
Heat of vaporization H_V	$MJ kg^{-1}$	0.37	Green and Perry (2008)		
Heat of combustion H_C	$MJ kg^{-1}$	45.1	Green and Perry (2008)		
Radiation fraction F_s	-	0.3^{*}	Mannan (2012)		
* concompting value accurated for the present each study					

Table A.3: Physical properties of n-hexane

^{*} conservative value, assumed for the present case study